

The logo for Helden Aerospace features a stylized red and black swoosh that frames the company name. The swoosh starts as a thick red shape on the left, curves around the top and bottom, and tapers to a thin black line on the right. The text "Helden Aerospace" is written in a bold, black, sans-serif font across the center of the swoosh.

Helden Aerospace

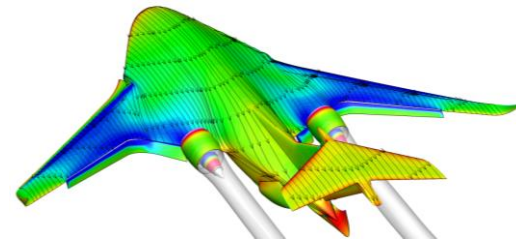
HeldenMesh Grids
Effect of Slat Bracket Shape on RANS CFD
Simulations for Case 1

Andrew Wick

Rick Hooker

Slat Bracket Shape Effect

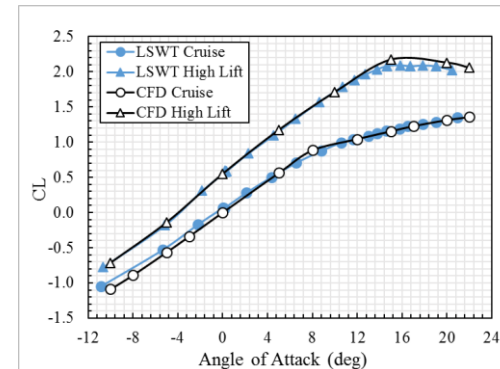
- The current HLPW wind tunnel bracket shapes are difficult for RANS
 - Accuracy: Pizza slice separation on wing upper surface which is not observed in testing are caused by brackets
 - Residual convergence: some solvers have trouble converging to machine zero when the brackets are present, but others don't.
 - Many factors contribute: grid, condition, FE vs. FV, other solver schemes
- Big Picture: How are aircraft high lift systems designed? (Lockheed)
 - **Conceptual design phase:** slat brackets are usually not defined. Slat and flaps hang out in space. Engineers vary gap, overhang, angle, etc to get best RANS-based performance predictions without brackets
 - The "no bracket" geometry becomes the nominal baseline case
 - These are often the performance estimates that size the airplane
 - **Preliminary design phase:** wind tunnel testing. Model needs WT brackets!
 - **Detailed design phase:** slat tracks and actuation are only considered once the wing and slats designs are finalized and the loads are known. Ideally, CFD analysis is used to verify the final actuators do not impact performance (CLmax)
- How should wind tunnel model slat brackets be designed?
 - 1. Replicate typical full scale # brackets, but sized by wind tunnel constraints -> leads to current CRM HL designs
 - 2. Go through bracket design effort and get performance as close to the no bracket case as possible (fewer brackets, stream aligned, aerodynamic cross section) -> naturally leads to more RANS friendly designs.



Hybrid Wing Body Conceptual Design Model w/ Flaps and Slats (no brackets)
*AIAA 2014-1285

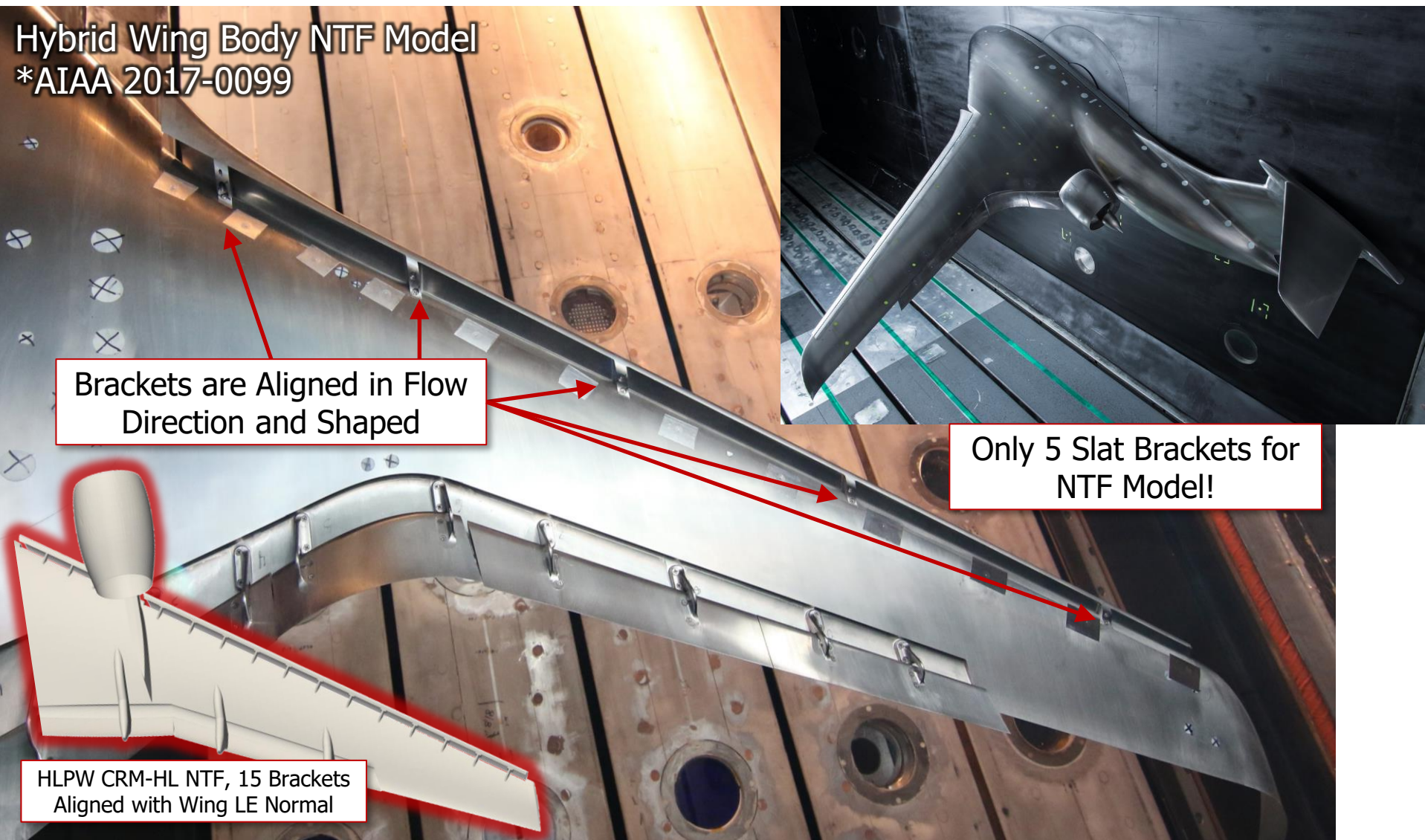


HWB LSWT Powered WT Model
*AIAA 2017-0100



RANS matches Test Data. No Pizza Sep!

NTF-HL Model Designed for Minimal Impact on Performance



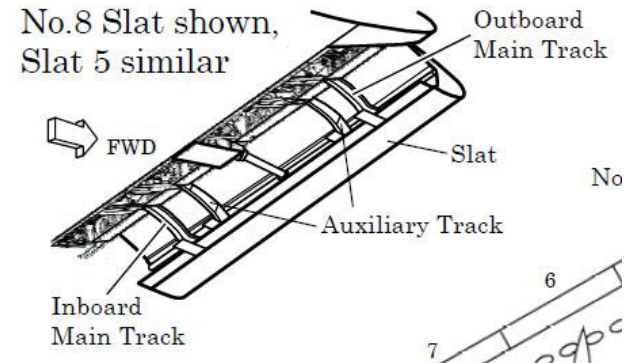
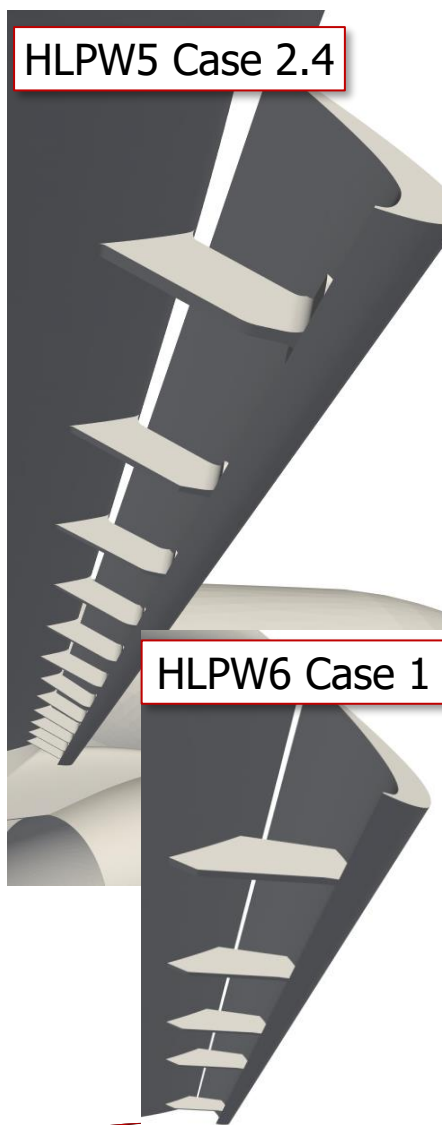
Hybrid Wing Body NTF Model
*AIAA 2017-0099

Brackets are Aligned in Flow Direction and Shaped

Only 5 Slat Brackets for NTF Model!

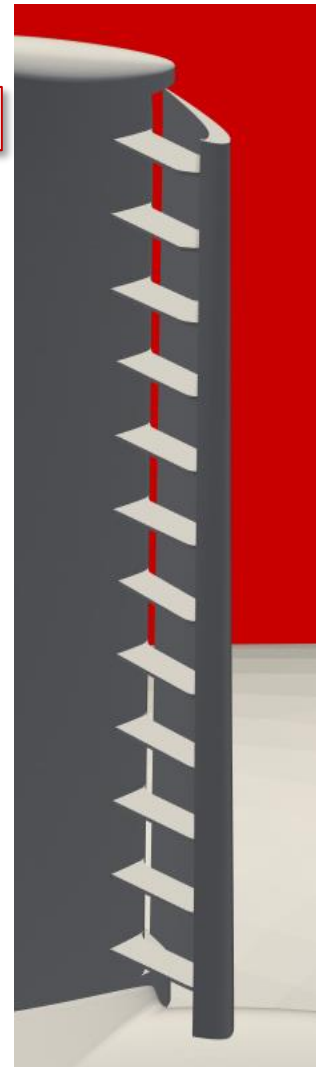
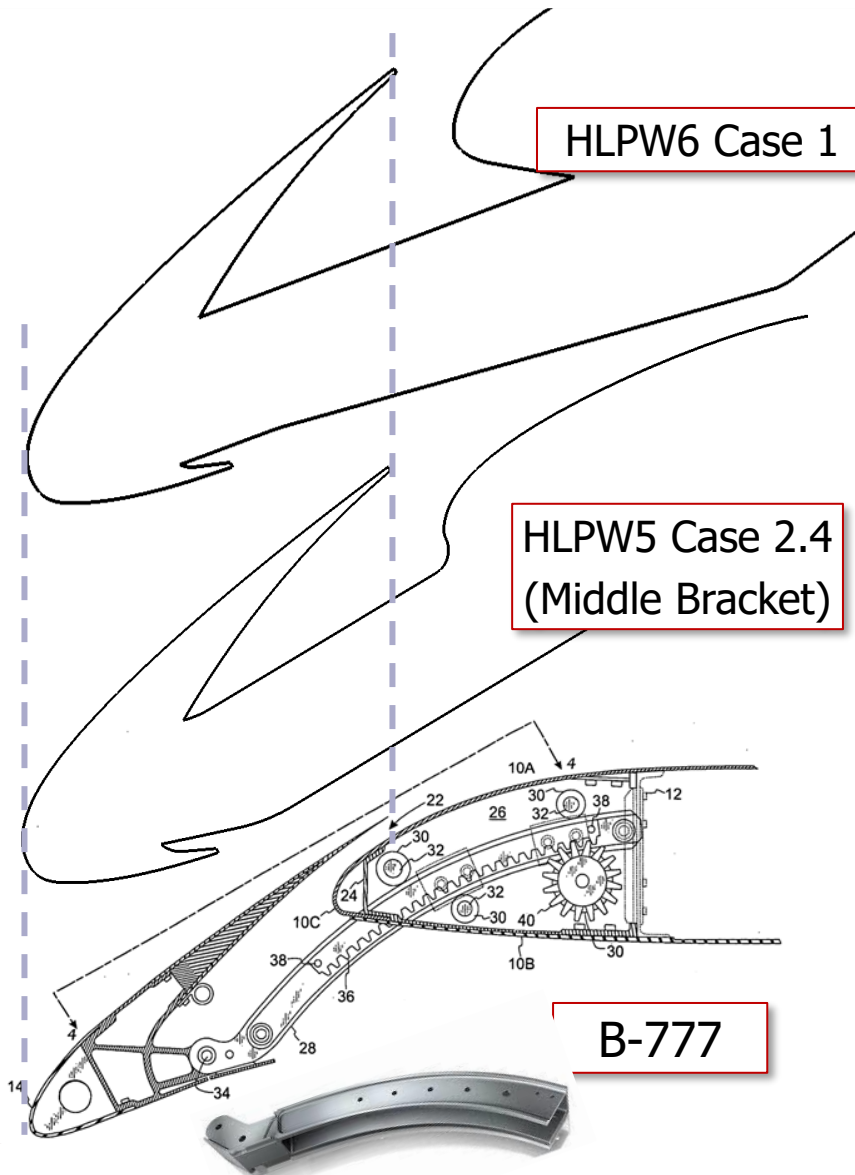
HLPW CRM-HL NTF, 15 Brackets Aligned with Wing LE Normal

Slat WT Brackets vs. Slat Tracks



<http://www.b737.org.uk/slat-track-ad.htm>

Slat WT Brackets vs. Slat Tracks



HLPW5 Case 2.4

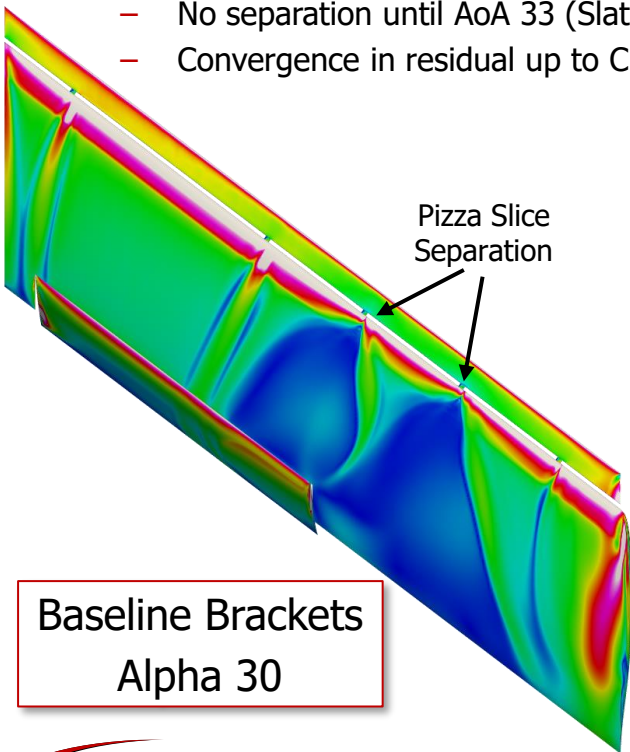
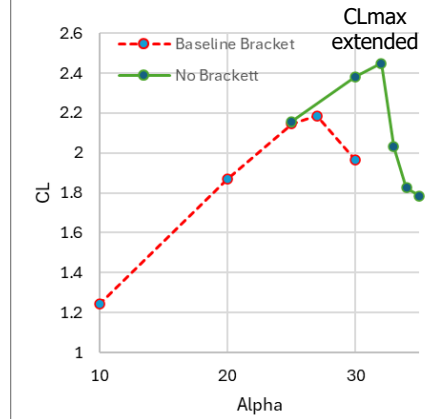
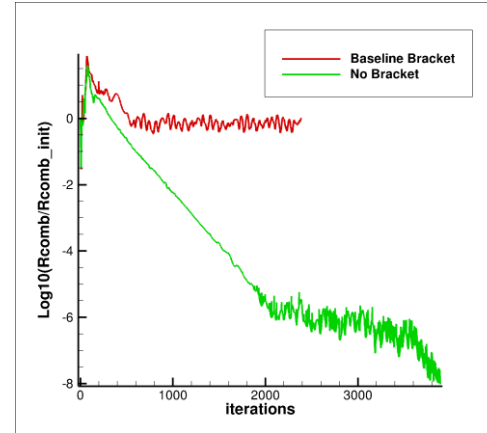


B-777

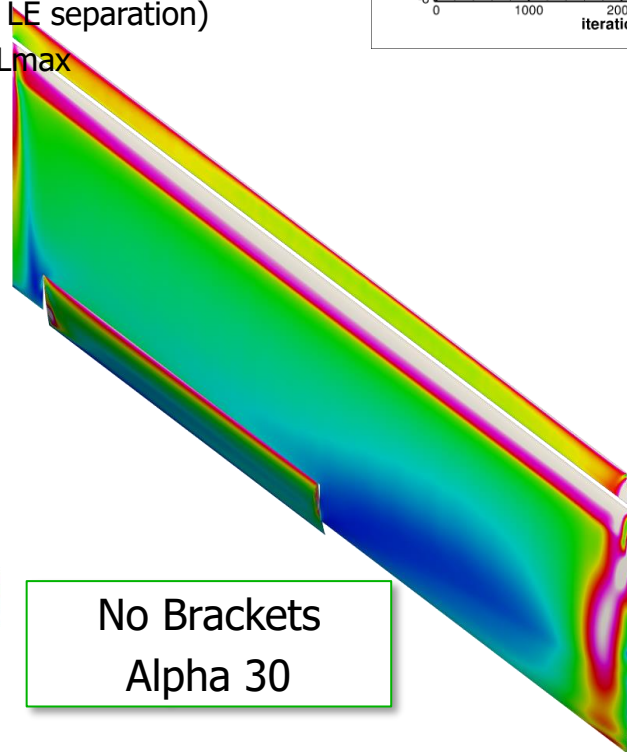
CRM Brackets Appear Larger / Chunkier Than Actual Slat Tracks

Case 1: No Brackets

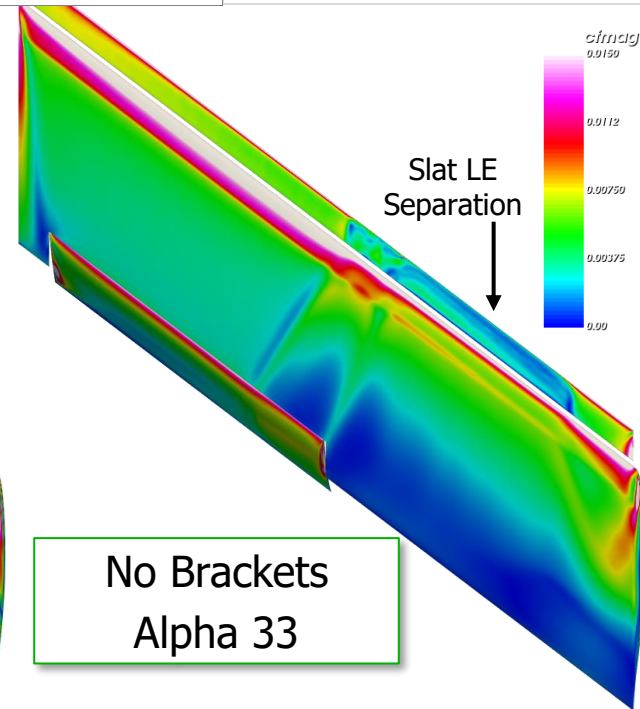
- HeldenMesh, USM3DME used for meshing/CFD
 - Series 3, D level fixed mesh spacing + wake sources
 - ~40M cells for each bracket variation
- Baseline bracket design impact on RANS
 - Residual convergence difficulties with HANIM
 - Pizza Slices on wing upper surface at AoA 30
- No bracket impact on RANS
 - No separation until AoA 33 (Slat LE separation)
 - Convergence in residual up to CLmax



Baseline Brackets
Alpha 30



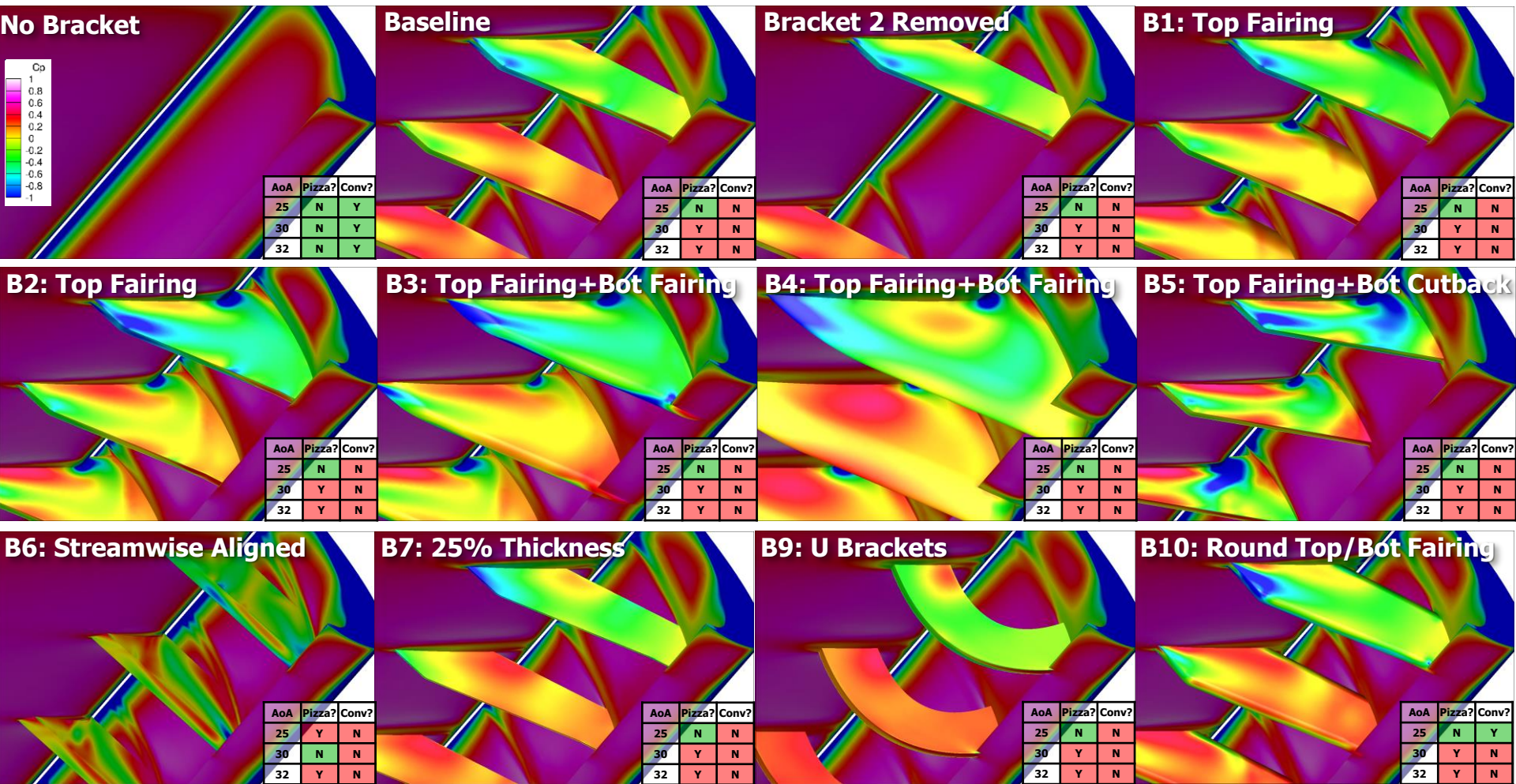
No Brackets
Alpha 30



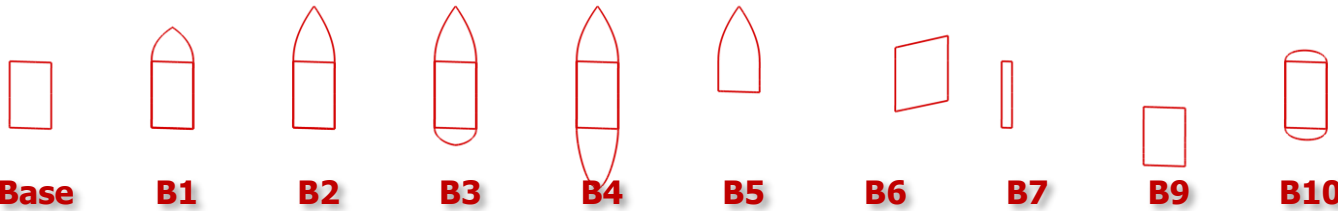
No Brackets
Alpha 33

No Bracket = Solution Convergence and no Pizza

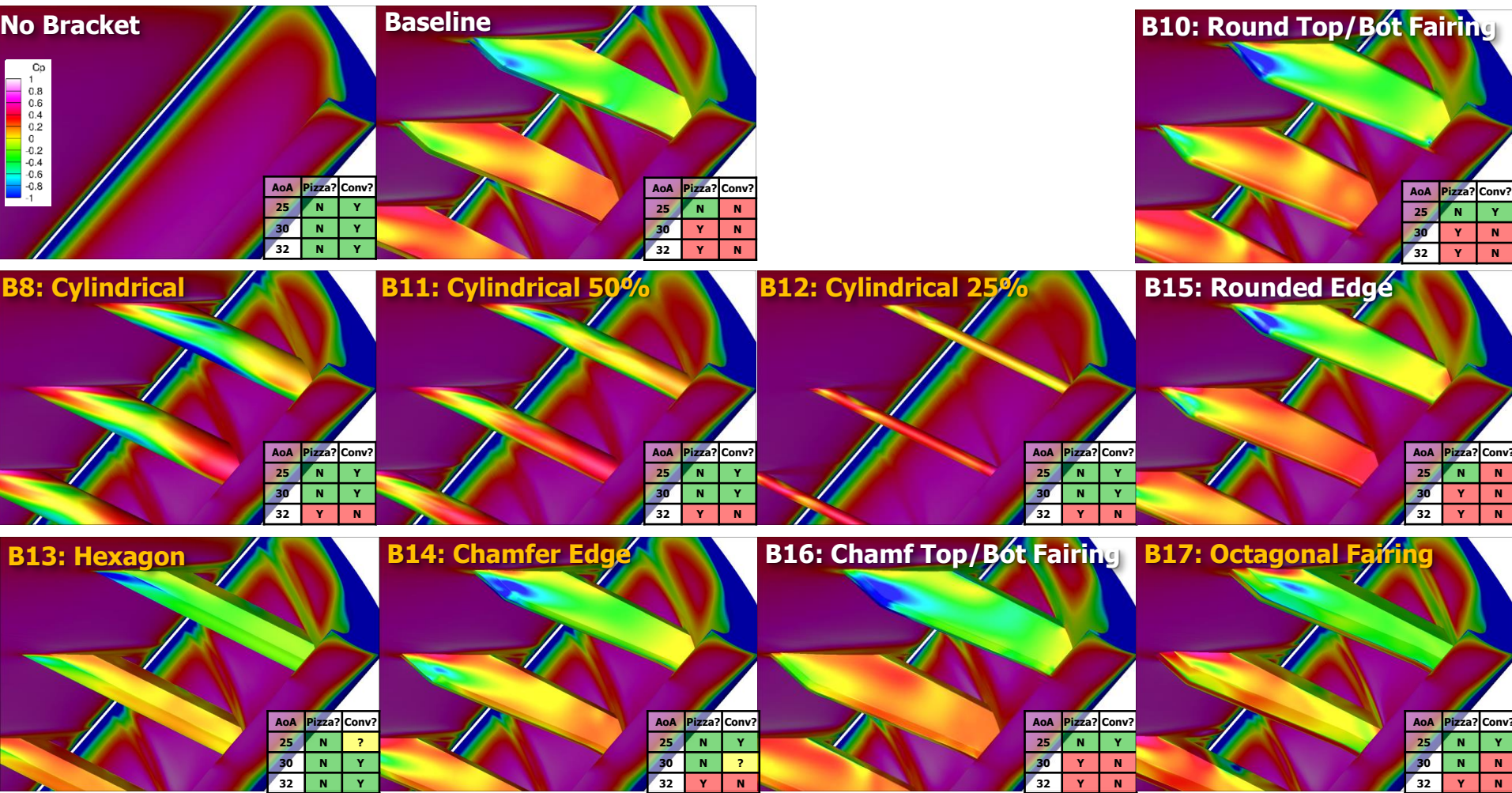
Bracket Design Effort (1/2)



Cross Sections of Brackets:



Bracket Design Effort (2/2)



Cross Sections of Brackets:

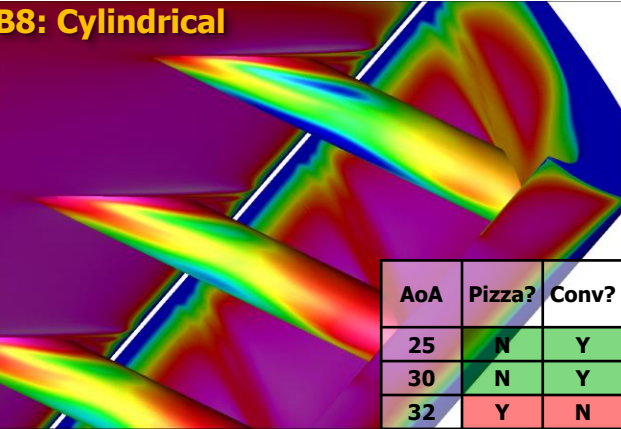


Base B10 B8 B11 B12 B15 B13 B14 B16 B17

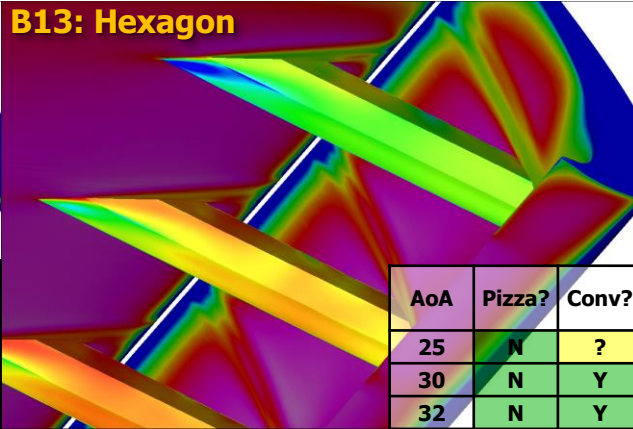
Bracket Study Summary



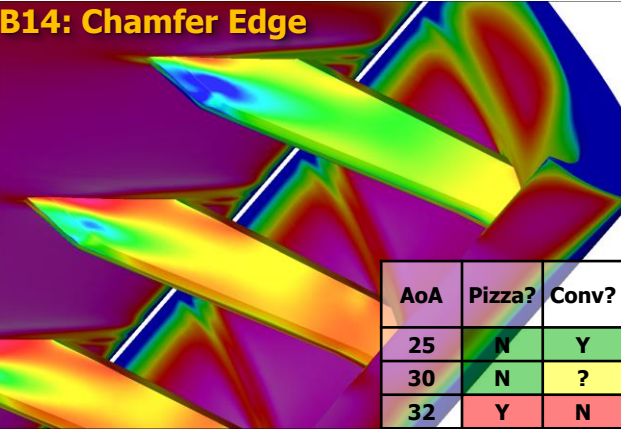
B8: Cylindrical



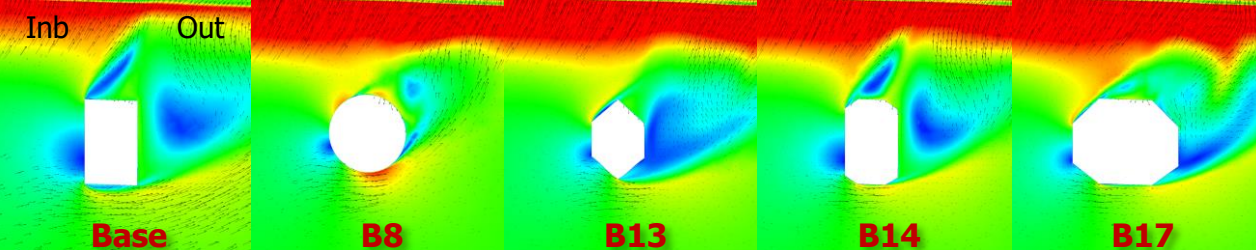
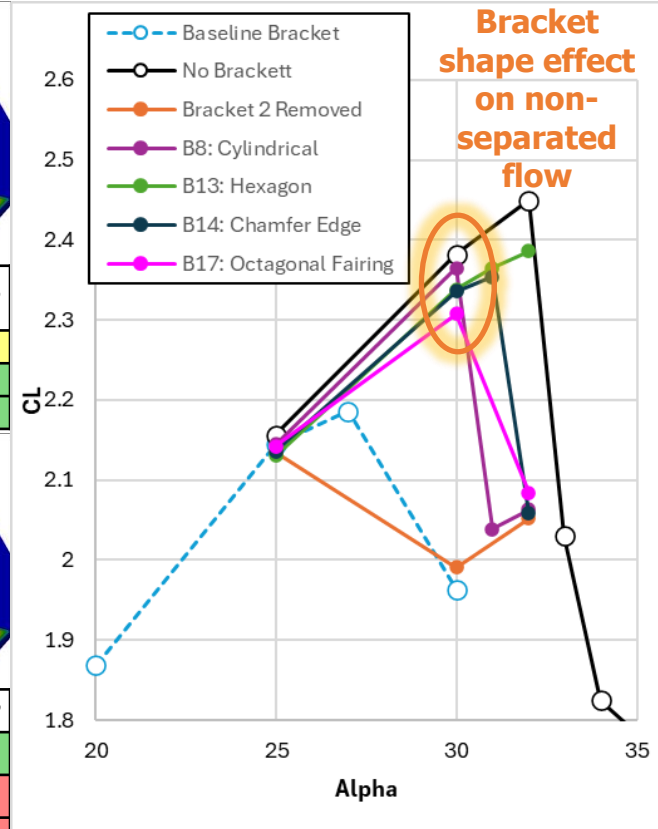
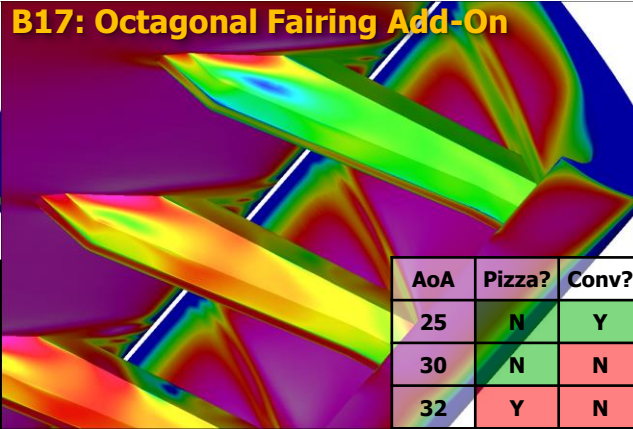
B13: Hexagon



B14: Chamfer Edge



B17: Octagonal Fairing Add-On



Cross Sections of Brackets:

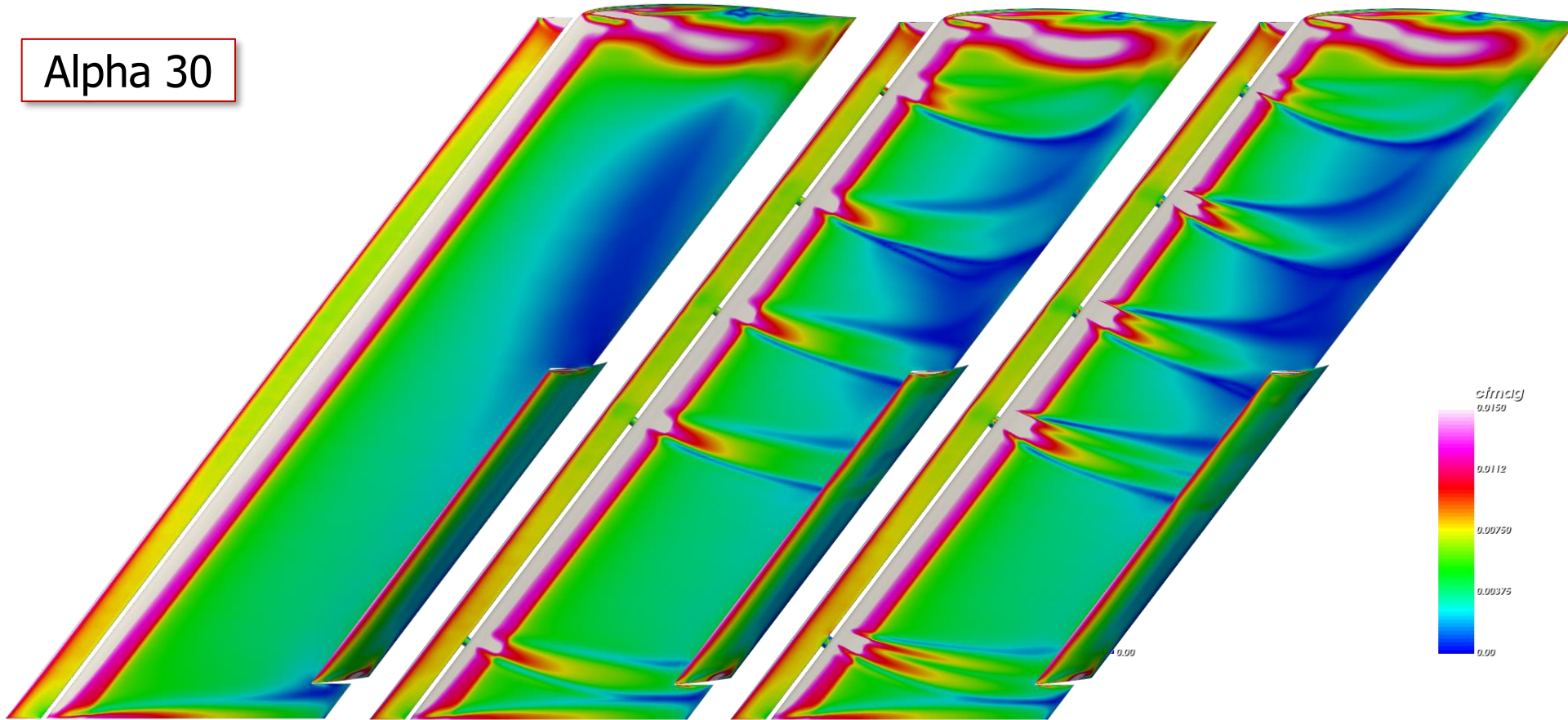


M 0 0.065 0.13 0.195

Bracket Study Conclusions



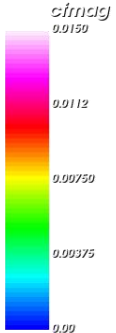
Alpha 30



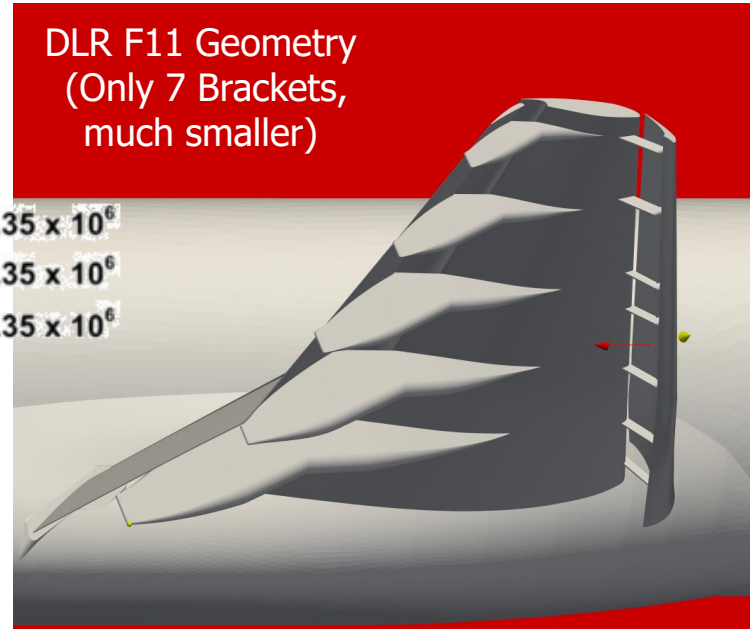
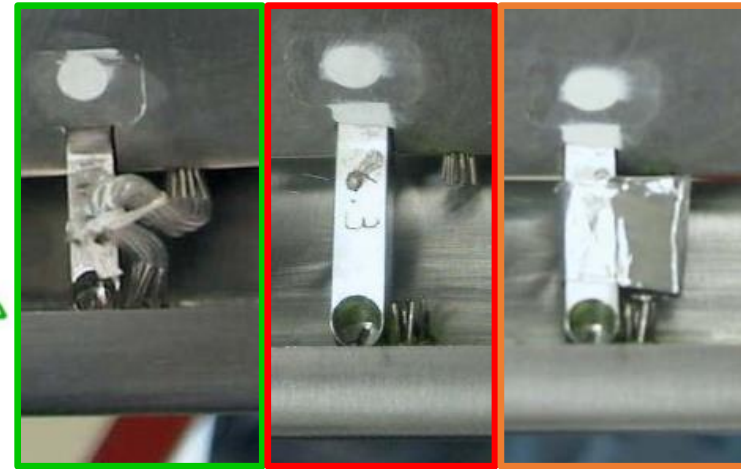
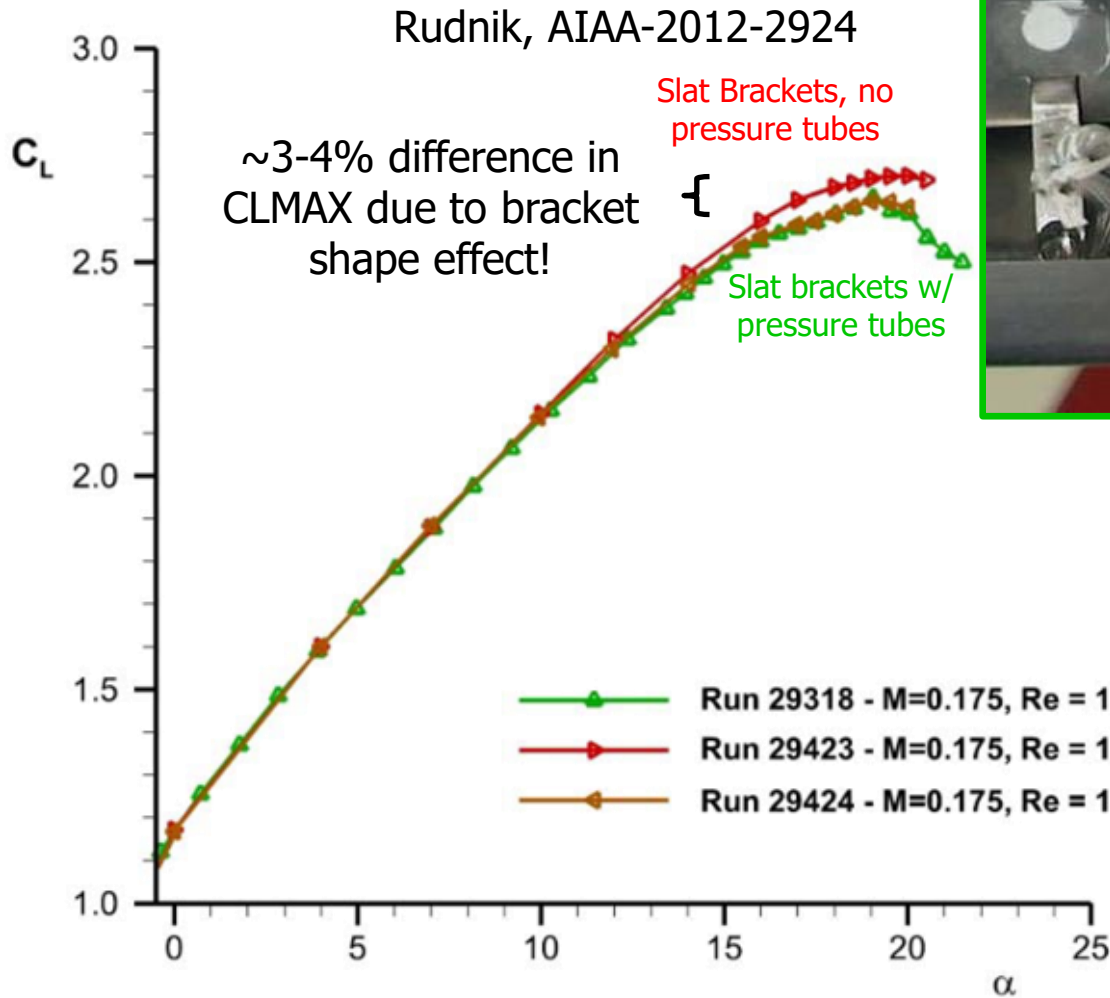
No Bracket
CL 2.382

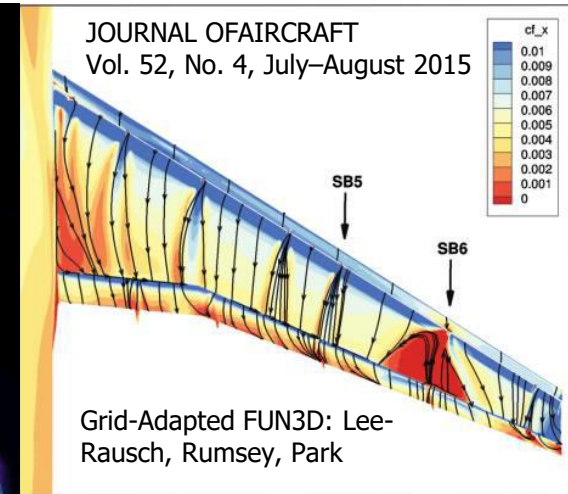
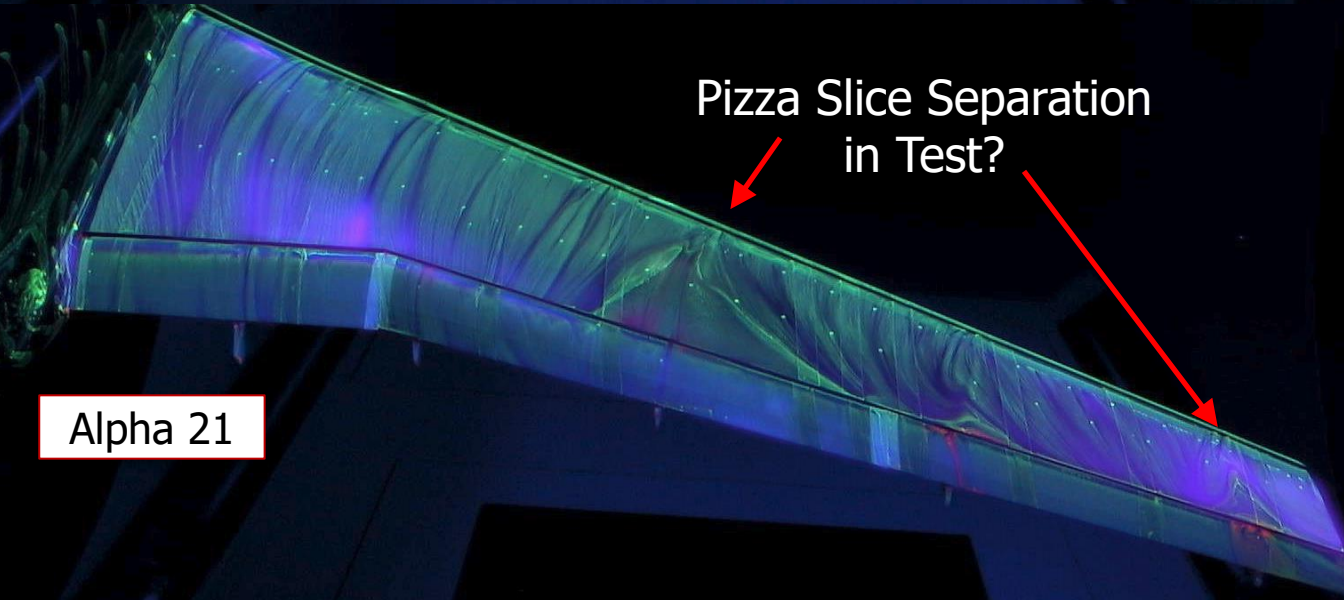
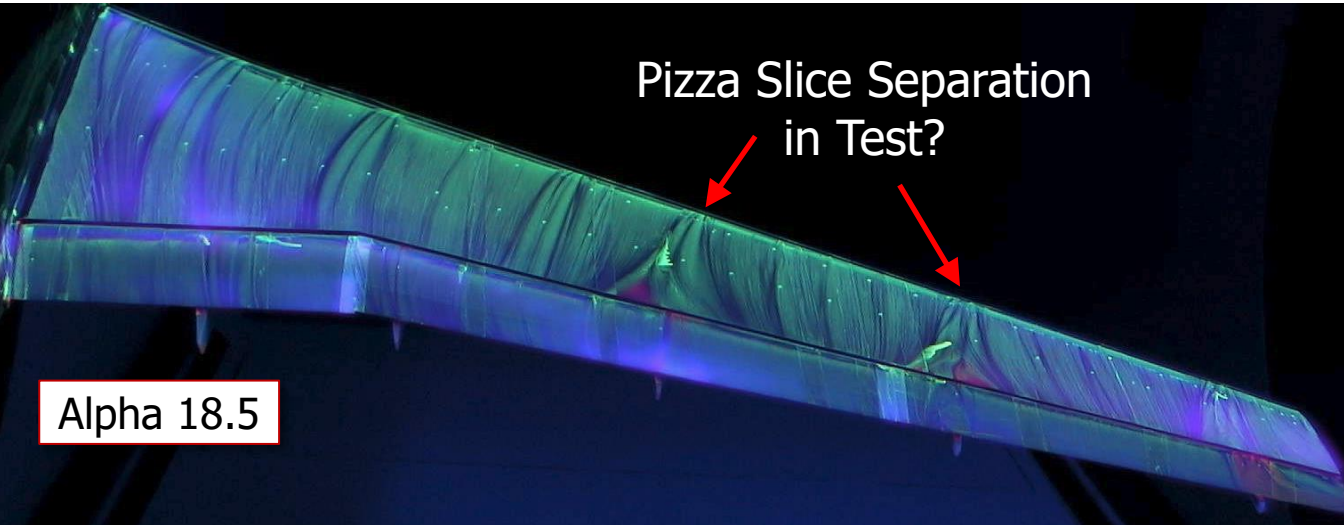
B8: Cylinder
CL 2.365

B17: Octagonal Fairing
CL 2.307



Bracket Shape Also Has Effect on CL for Attached Flows Near CLMAX

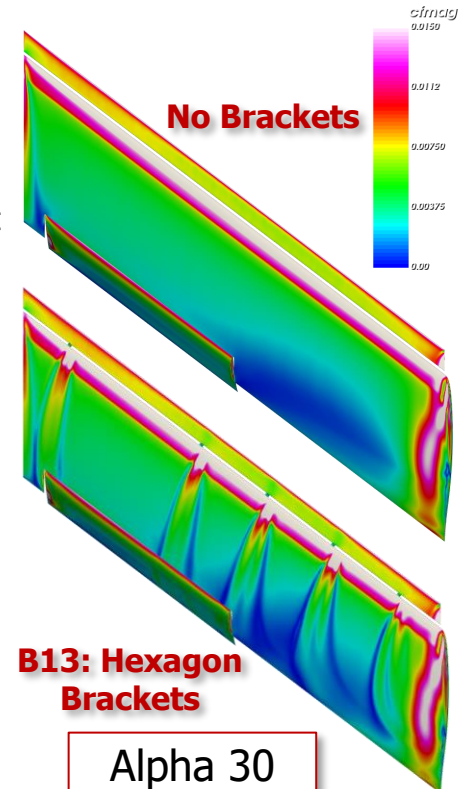
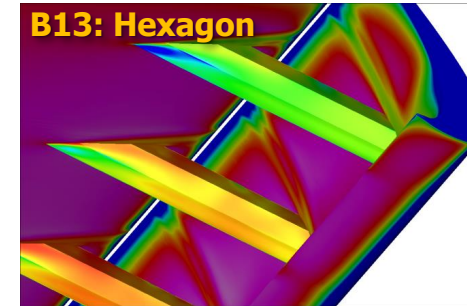




"In my opinion, pizza separation probably was an issue for RANS in HLPW2, but there was no collection of surface streamlines back then, so it's hard to pin it down for all of the collective results. We just weren't on the lookout for that. However, some participants' papers probably showed it." –Chris Rumsey (May 2026)

How would we do on this case today?

- Some minor adjustments to the bracket shape can result in improved RANS simulations with no pizza and improved residual convergence
 - Cylindrical cross section
 - Hexagon cross section
 - Chamfering on sharp edges of existing brackets.
 - Fairings can be added to the sides of the existing brackets.
 - These changes might also improve CLmax on the test data.
- The “No Bracket” Case is still useful to study, even if it can’t be tested
 - Simulates the nominal performance of the high lift system. It’s what we designed!
 - In the aircraft design process, most pre-test RANS CFD probably doesn’t even model slat brackets. These are the performance estimates that size the aircraft!
 - Easier to simulate with RANS. Free of pizza and convergence issues.
 - For HLPW5 Case 2, RANS simulations without brackets matched the test data
- Where do we go next on this topic?
 - Look at Case 1 with alternate bracket shape (with or without test data)
 - Look at Case 1 with no brackets
 - Revisit HLPW2: What effect do the smaller brackets have on our scatter between participants? How do we compare to a case that actually shows pizza-like separation as the stall mechanism?
 - There was a 10% CRM test with alternate “aero” and “acoustic” brackets from NASA Langley that may be interesting...

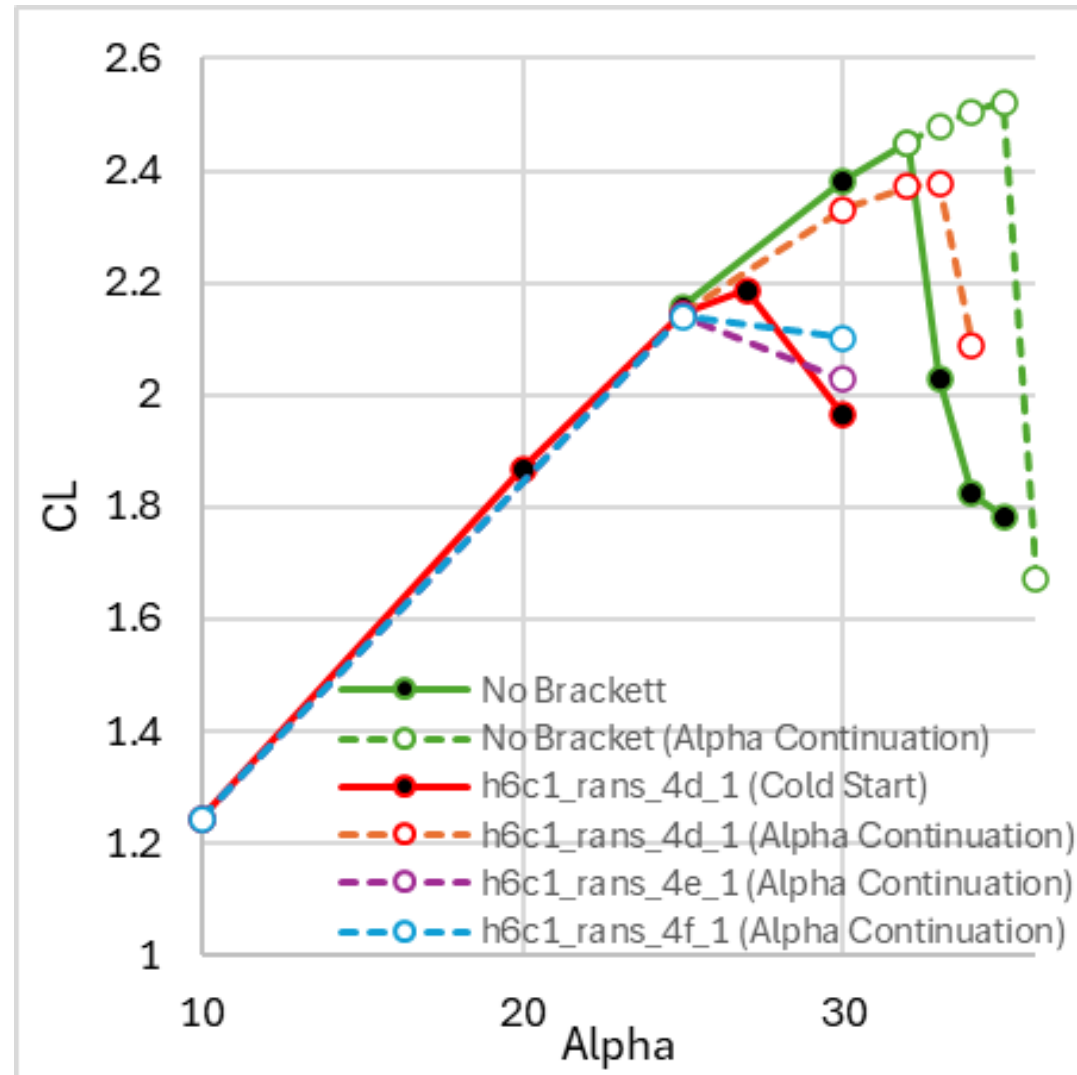


Backup



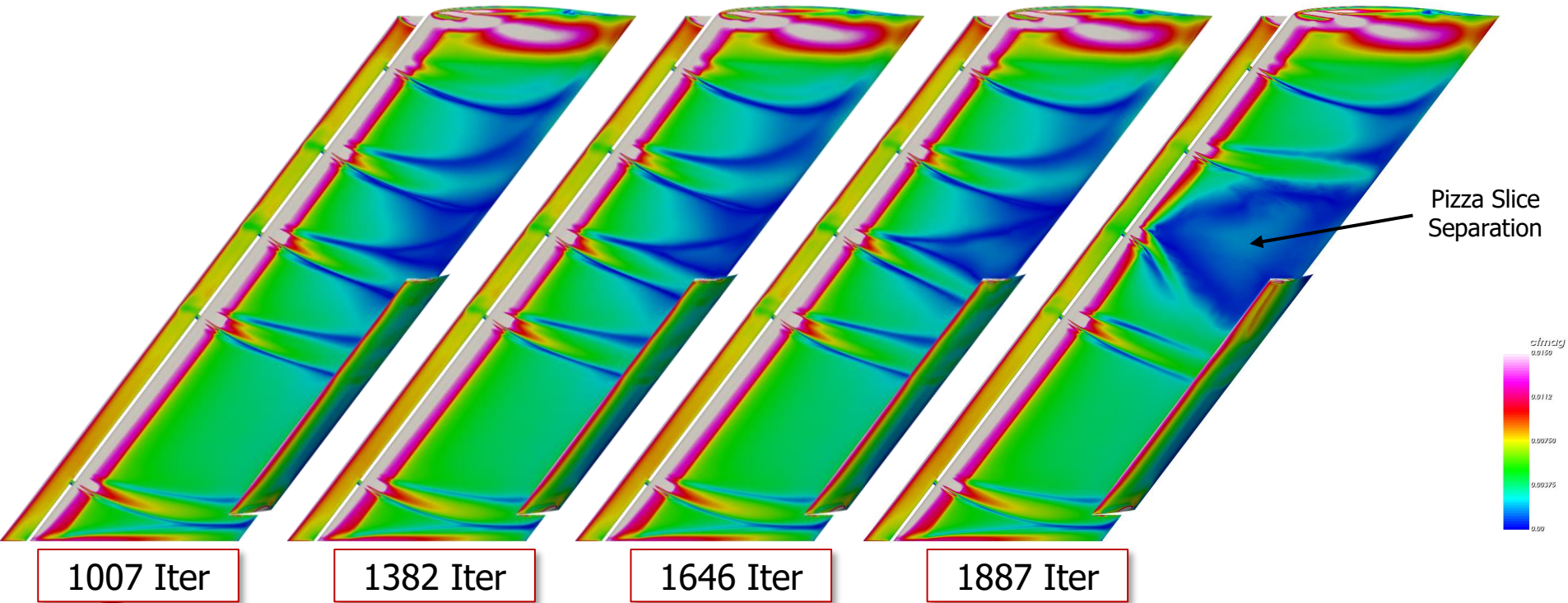
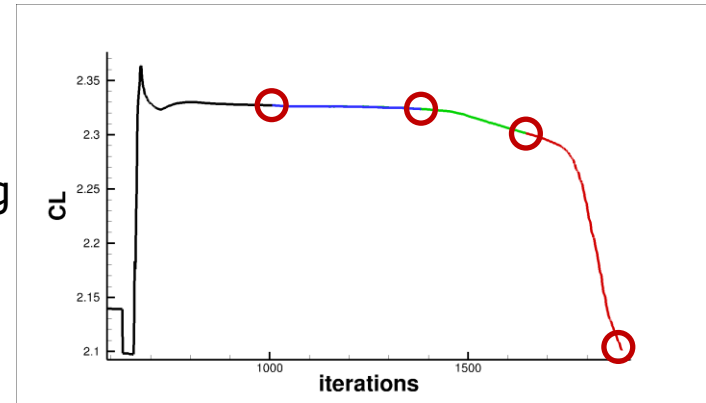
Effect of Alpha Continuation

- Effect of using alpha continuation was explored. Alpha 30 simulation was restarted from the pizza-free Alpha 25 deg solution
- Alpha continuation delays stall by 3 deg AoA for the no bracket simulation
 - C_{lmax} at Alpha 32 \rightarrow C_{lmax} to Alpha 35
- For the medium mesh (4D), alpha continuation can be used to get pizza-free solutions at alpha 30
- Finer mesh resolution (4E, 4F) leads to development of pizza separation
- Explorations were also made using heldenadapt to create adapted meshes which were tailored to the pizza-free Alpha 30 simulation. These simulations all led to pizza separation eventually forming
 - Alpha continuation used to simulate alpha 25 deg then restarted at alpha 30. As mesh was refined, pizza eventually formed



Bracket Study

- h6c1_rans_4f_1 mesh (323M cells) was used to study development of the Pizza Separation at Alpha 30
- Solution was restarted from alpha 25 solution up to 30 deg
- Flow is initially pizza free. Then the 3rd bracket wake begins to widen at the trailing edge and eventually separates.



Bracket Study

- Alpha 30 Comparison of Bracket Shape Effect on Bracket Wakes
- B13 solution is grid converged using heldenadapt

