

xRANS – An Engineering Approach to Extend Flow Analysis Using RANS-Simulations

Stefan Melber-Wilkending / Cord Rossow / Ralf Rudnik / Niko Bier
Institute of Aerodynamics and Flow Technology
Transport Aircraft Department
HighLift Group
German Aerospace Center (DLR)

A large, curved image of the Earth from space occupies the bottom right portion of the slide. It shows a view of the planet's surface with blue oceans, green landmasses, and white clouds. The curvature of the Earth is prominent, and the image is positioned as if looking down from a high altitude.

Wissen für Morgen

Contents

- Introduction
- Analysis of RANS / uRANS / hybrid RANS/LES
- Development of the xRANS method
- Demonstration of xRANS-Method, Results and Discussion
- Conclusion

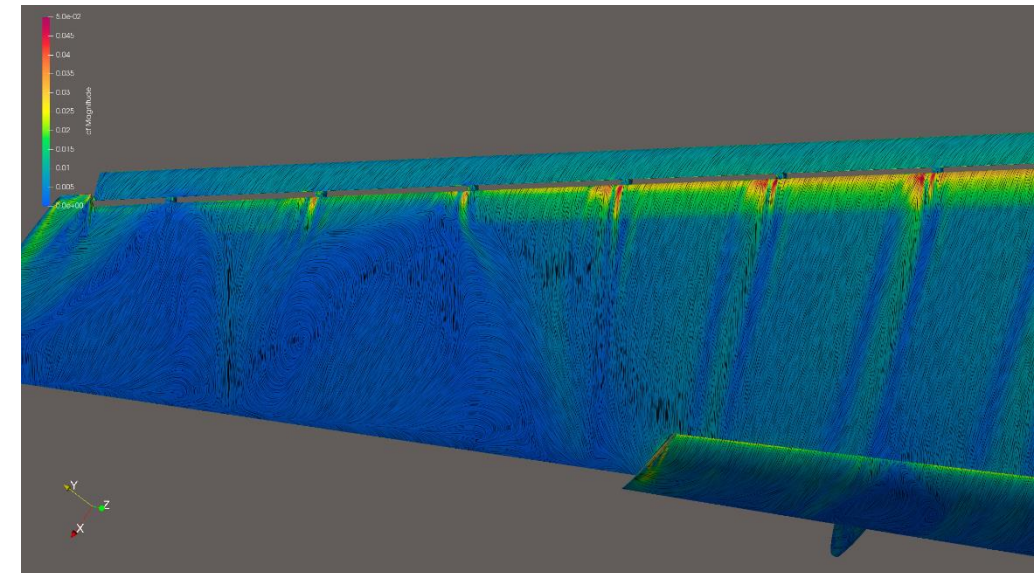
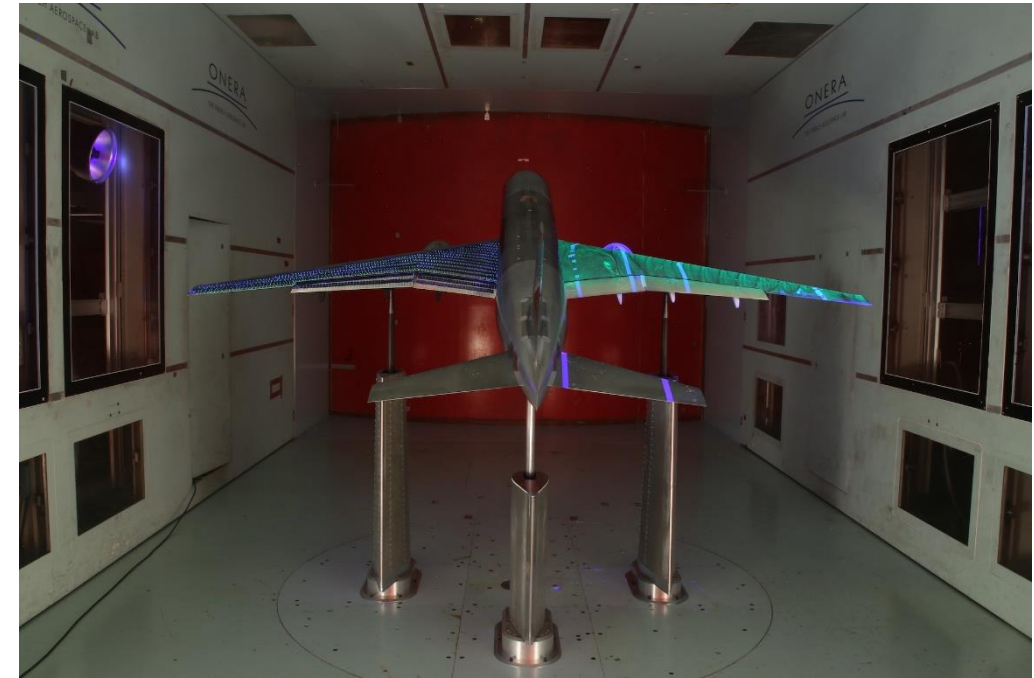


Introduction



Introduction

- 4th High Lift Prediction Workshop: Common Research Model – High Lift (CRM-HL) → major simulation challenge for RANS methods:
“pizza-slice” separation:
 - lambda-shaped separation areas
 - triggered by the wake of the slat brackets
 - develop towards the trailing edge
 - expanding in size with increasing angle of attack
 - separations occur significantly earlier (AOA) compared to experiments → premature lift breakdown → inconsistent simulation of the lift breakdown e.g. for the CRM-HL
 - onset tends to occur closely downstream of the brackets & overpredicted in size → not reflected in experimental evidence
 - “pizza slice” sep.: only used for overpredicted separation areas
 - lambda-shaped separation: (measured) separation phenomena
- Starting point of this project: hybridRANS/LES are in very good agreement with experiments → compare RANS, uRANS & hybridRANS/LES simulation results in the region of “pizza slice separation” → insight in differences & identification of possible reasons

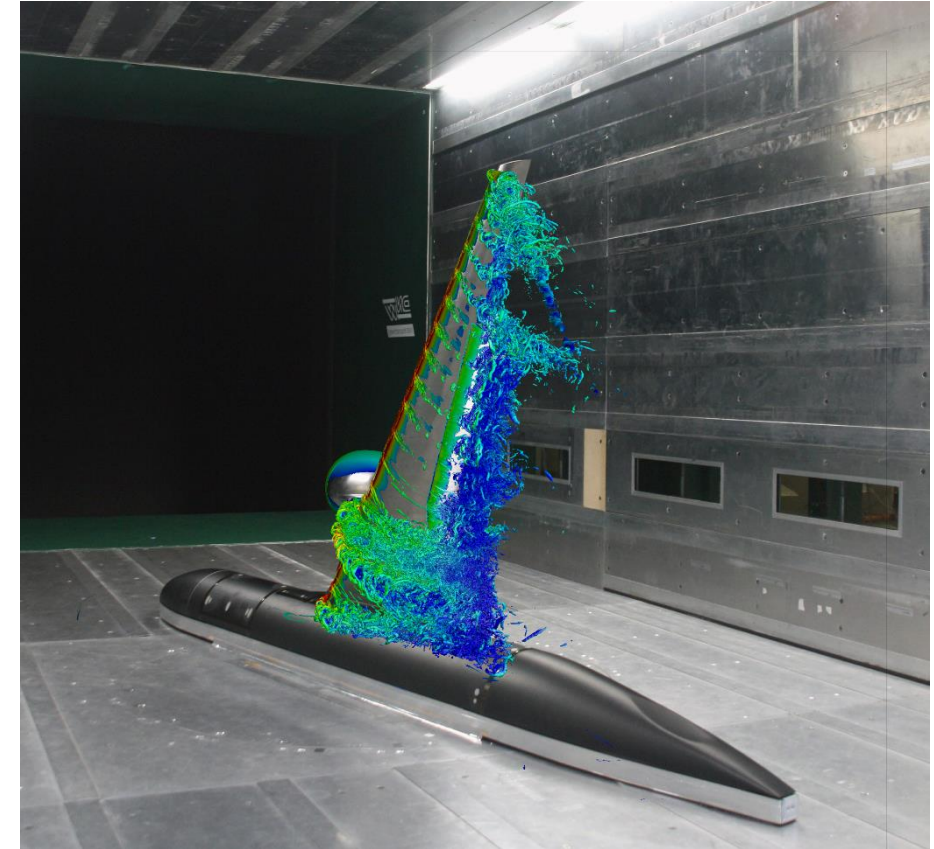


Analysis of RANS / uRANS / hybrid RANS/LES



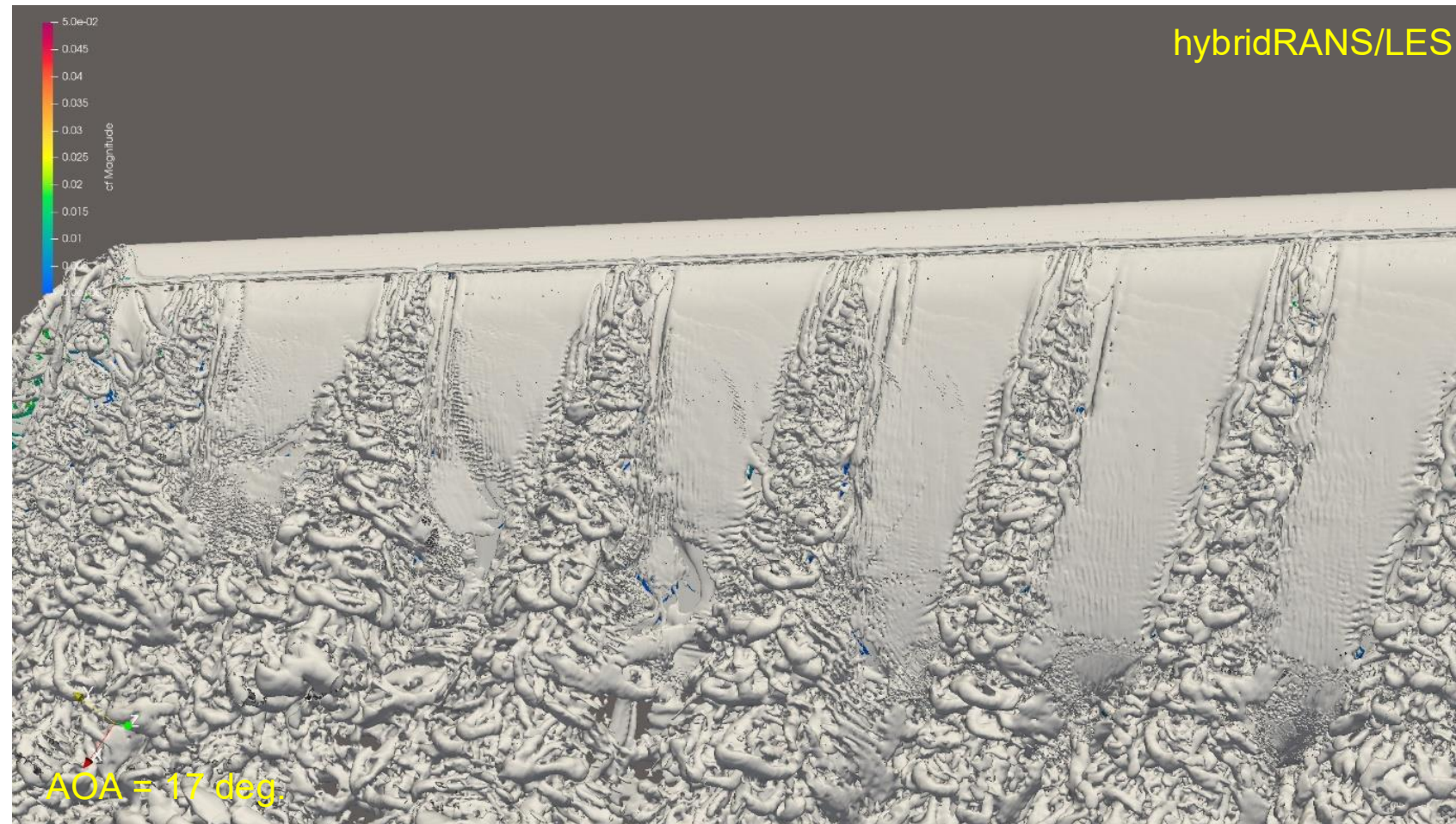
Analysis of RANS / uRANS / hybrid RANS/LES

- Basis for the presented numerical results: Freeflight simulation of CRM-HL (complete aircraft):
 - DLR-Project Adamant (internal)
 - DLR TAU-Code – RANS / uRANS / hybridRANS/LES
 - AOA: 17 deg. / Mach number: 0.2 / Reynolds number: $1.7e6$
 - SANeg turbulence model
 - uRANS & hybridRANS/LES → started from RANS solutions at the same AOA (including possible pizza slice separation)
 - ANSA quad / hexa dominant meshes with resolution “fine”
- Different mesh resolutions:
 - RANS: 296e6 points
 - uRANS: 215e6 points & normalized time step size: 0.001
 - hybridRANS/LES: 548e6 points & normalized time step size: 0.0004



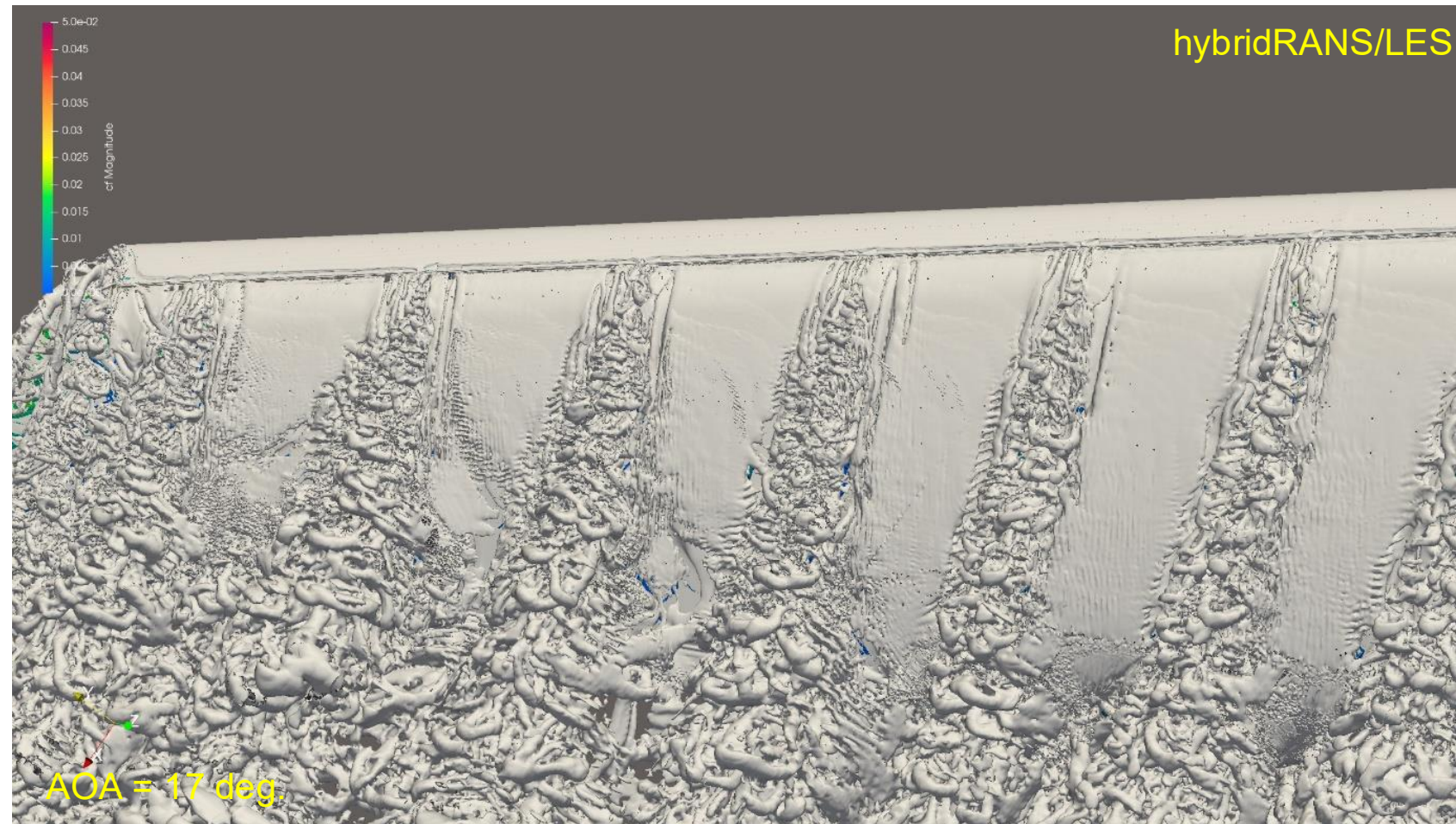
Analysis of RANS / uRANS / hybrid RANS/LES

- hybridRANS/LES simulations proven for very good agreement with experimental results even in the region of highly separated flow → trusted baseline to compare uRANS & RANS with
- Visualization: lambda2 criterion
- Two main flow features:
 - small-scale turbulent structures of wake-type behind blunt elements in in region of pressure rise
 - vortices (mostly) aligned with the main flow



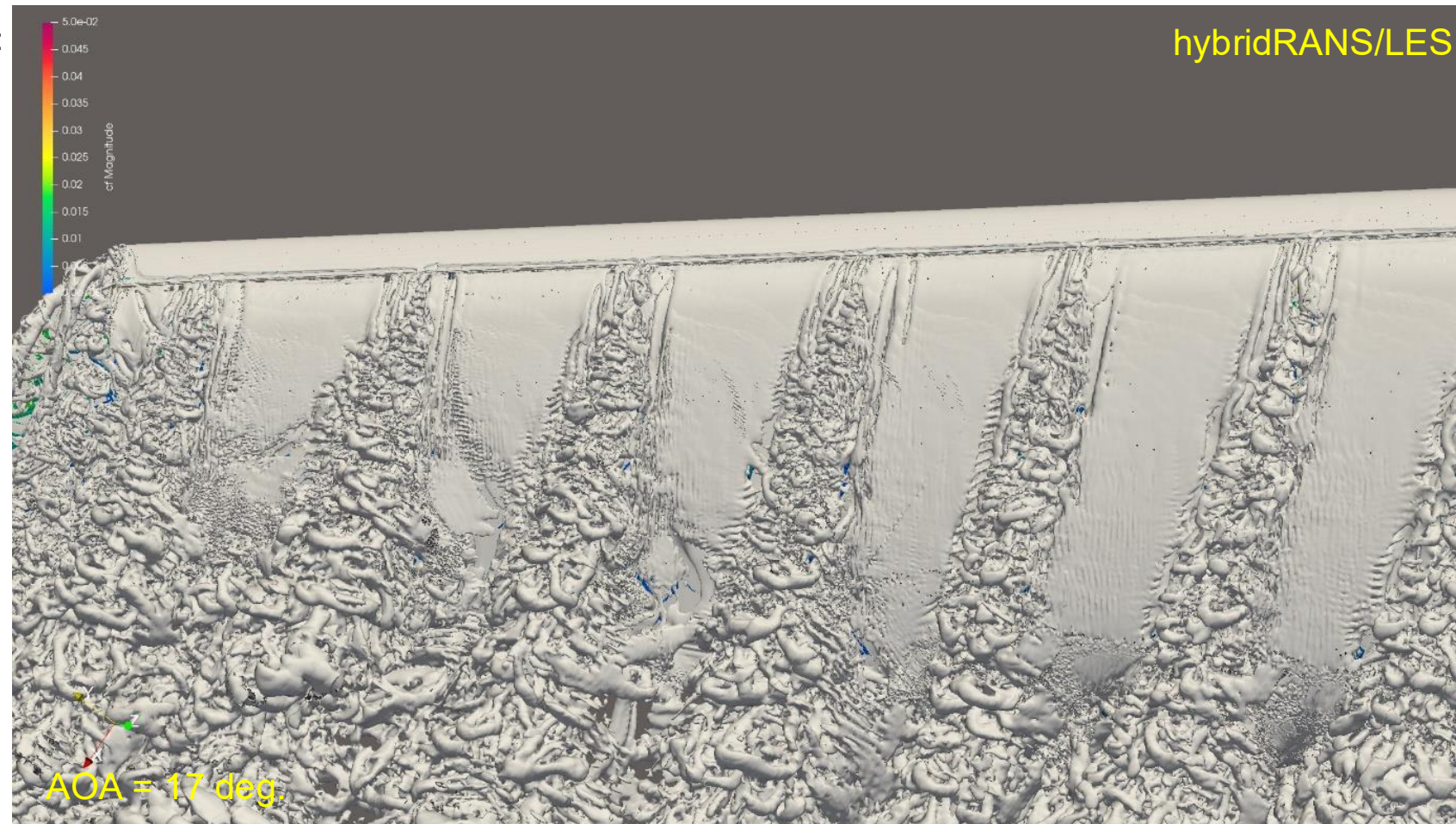
Analysis of RANS / uRANS / hybrid RANS/LES

- Primary and secondary horse shoe vortex around the slat brackets – early break-down in turbulent flowfield (compared to RANS)
- Slat cavity vortex:
 - driven by viscous forces of the flow through the slat gap
 - disturbed by the slat brackets → partially turned into flow direction
 - both for the inboard- and outboard side of a slat bracket, because vortices have no start or end in the flow field (Helmholtz's theorems)
- All vortices partially interact in the region of the slat bracket
- Depending on AOA & pressure rise on the upper wing surface → possible vortex breakdown due to pressure rise or due to turbulent structures
- All these “features” add an extra load on the boundary layer on the upper wing & additional pressure rise on the outboard wing because of “missing” flap



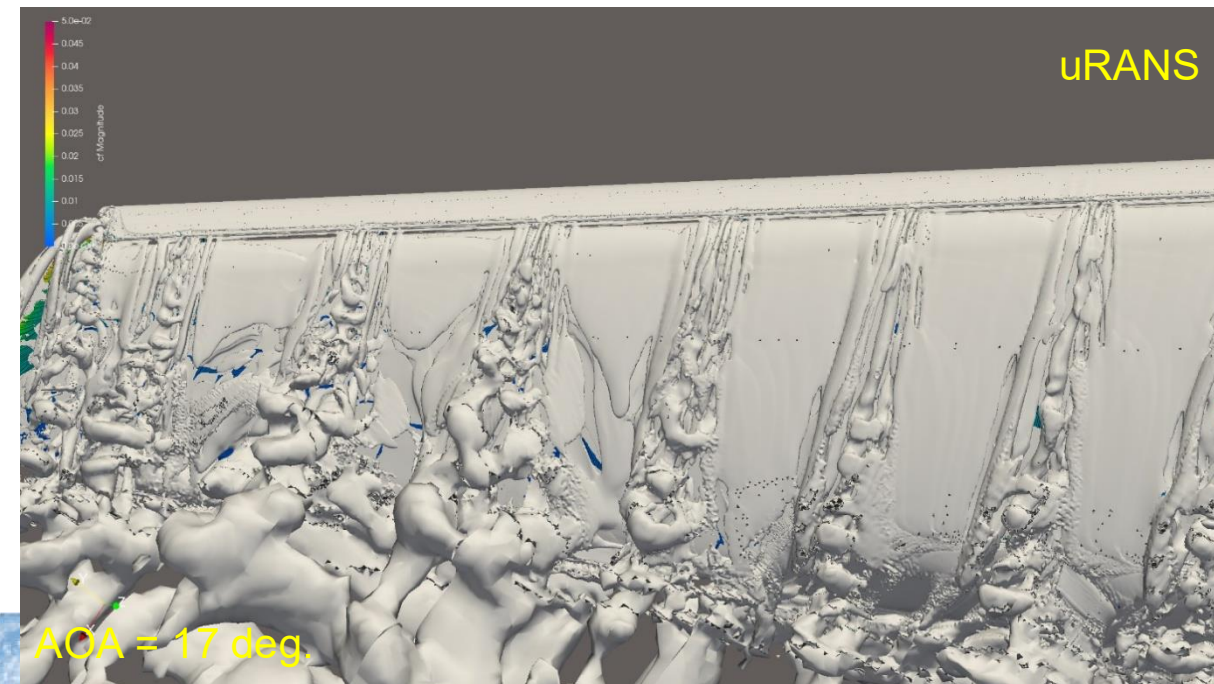
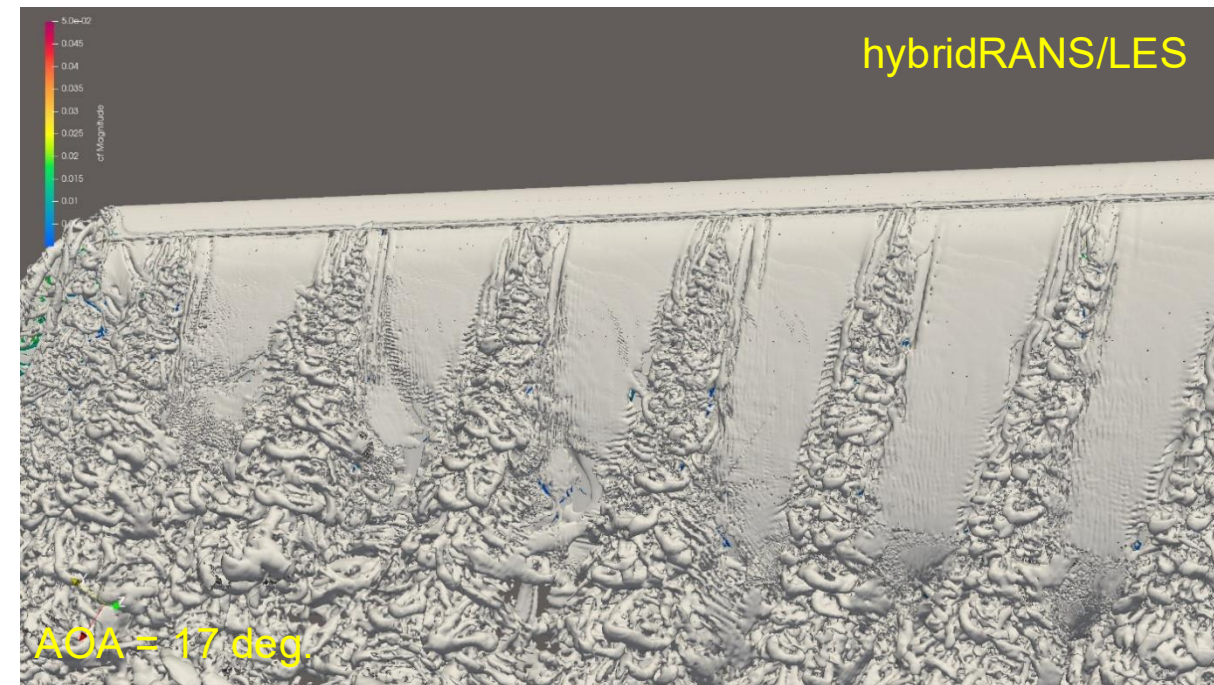
Analysis of RANS / uRANS / hybrid RANS/LES

- Local flow breakdown with increasing AOA:
 - local flow separation on the trailing edge (lambda-shaped) → grows upstream & spanwise direction till it reaches finally the slat bracket itself
 - Process is not abrupt in time & divides itself over some broader AOA range
- No clear distinction between turbulent flow field & vortices & fully developed flow separation in this region → trailing edge separation grows into the turbulent region against the flow direction
- The main flow passes through this region and transports the turbulent features downstream
- Time dependend flow - well confirmed by experiments (e.g. oil flow pictures, tufts, PIV, model vibrations at higher AOA...)



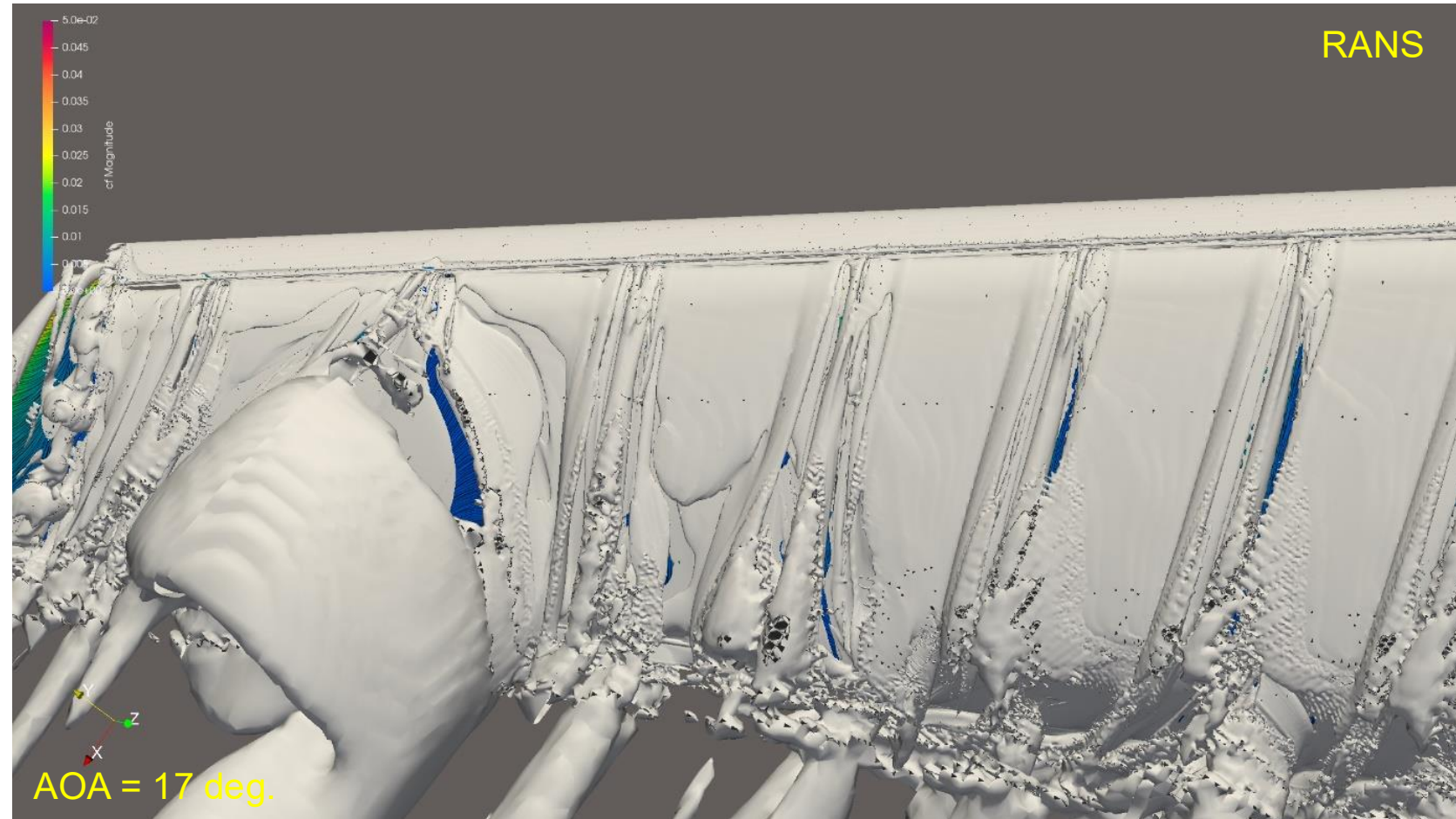
Analysis of RANS / uRANS / hybrid RANS/LES

- Reduced resolution in time step ($\sim$ factor 2.5) and spatial resolution ($\sim$ factor 2.0) compared to hybridRANS/LES $\rightarrow$ resulting turbulent structures larger in size
- Overall: flow topology & the lift coefficient compares well with the hybridRANS/LES & experiments $\rightarrow$ reduction in turbulence modelling fidelity did not change the overall behavior of the flow
- Remark: “best practise uRANS”:
 - Adequate temporal and spatial resolution (remark: example already “on the edge”)
 - Start from an “physical” RANS (e.g. no “Pizza slices”) – effect shown later in lift curve (17 deg ok, 19 & 21 deg not)
 - Dual time stepping: convergence of inner iterations at least one order of magnitude, two orders are typically fine (if not converging, reduce time step)
 - “Enough” CTUs $\rightarrow$ trustable mean value



Analysis of RANS / uRANS / hybrid RANS/LES

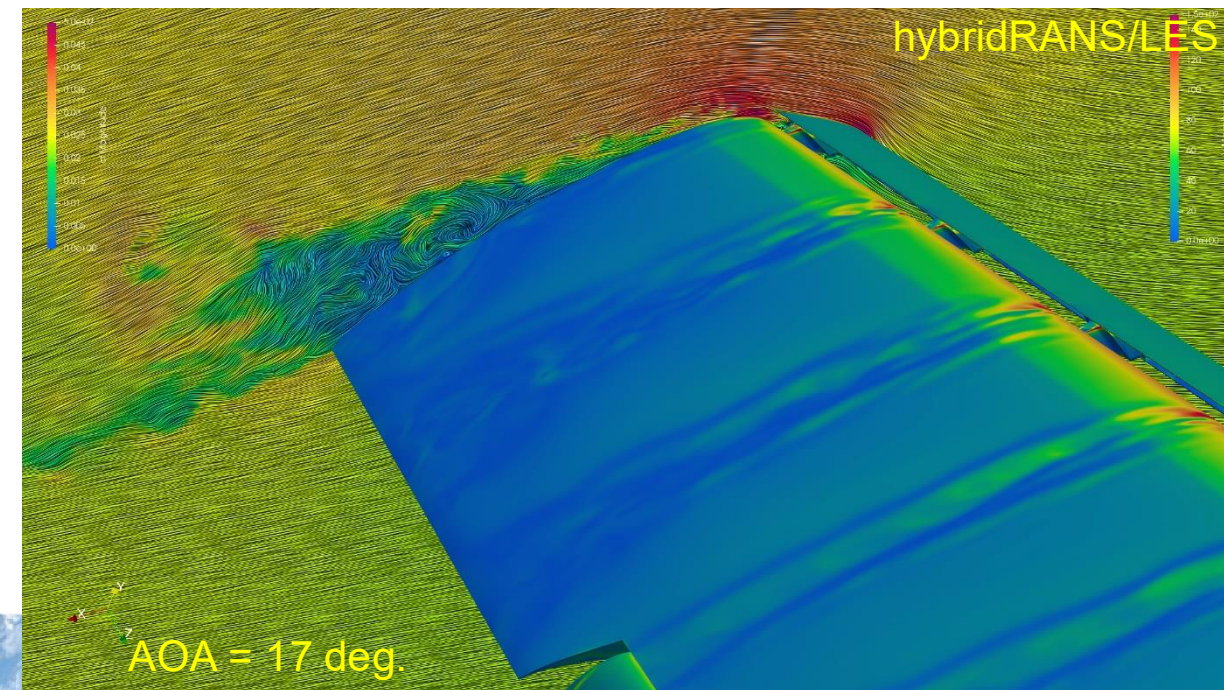
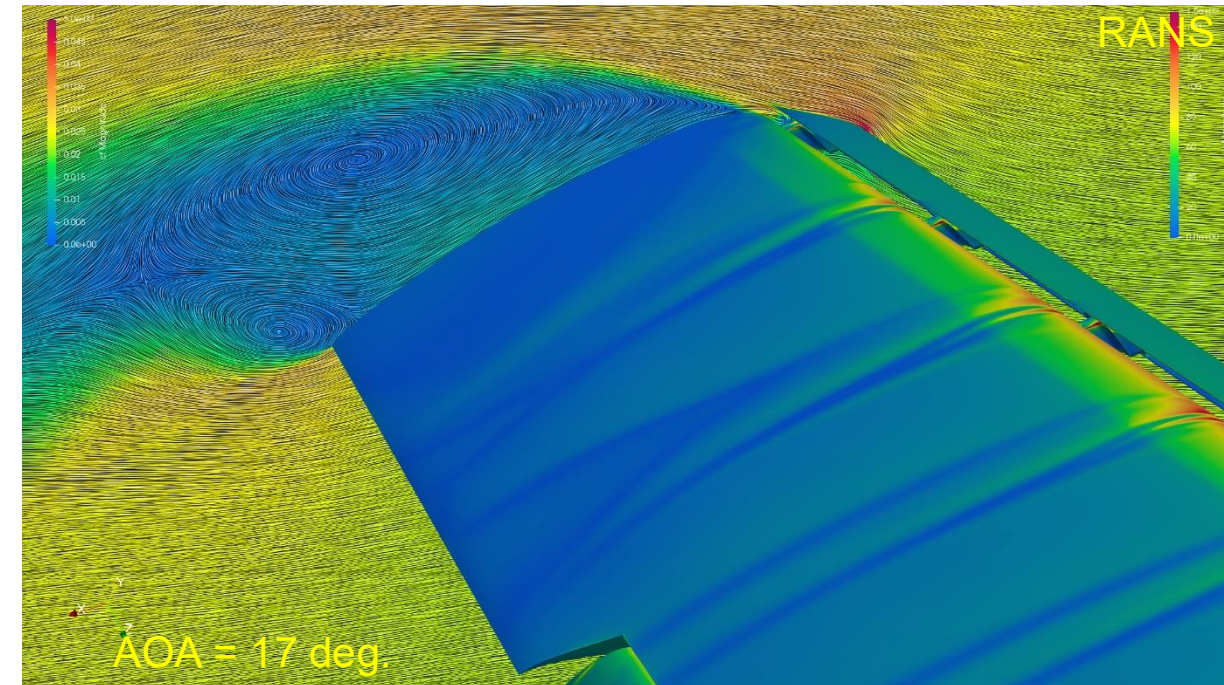
- Sharply marked flow separation region:
 - “pizza slice” shaped separation
 - nearly no “turbulent” structures, but kind a monolithic region in the flow
- starts on the trailing edge of the main wing element & with increasing AOA it grows upstream to the slat bracket
- bistable behavior: RANS solutions tend to show only not-separated or fully separated flow on the upper wing behind the slat brackets (“upper” and “lower” branch)



Analysis of RANS / uRANS / hybrid RANS/LES

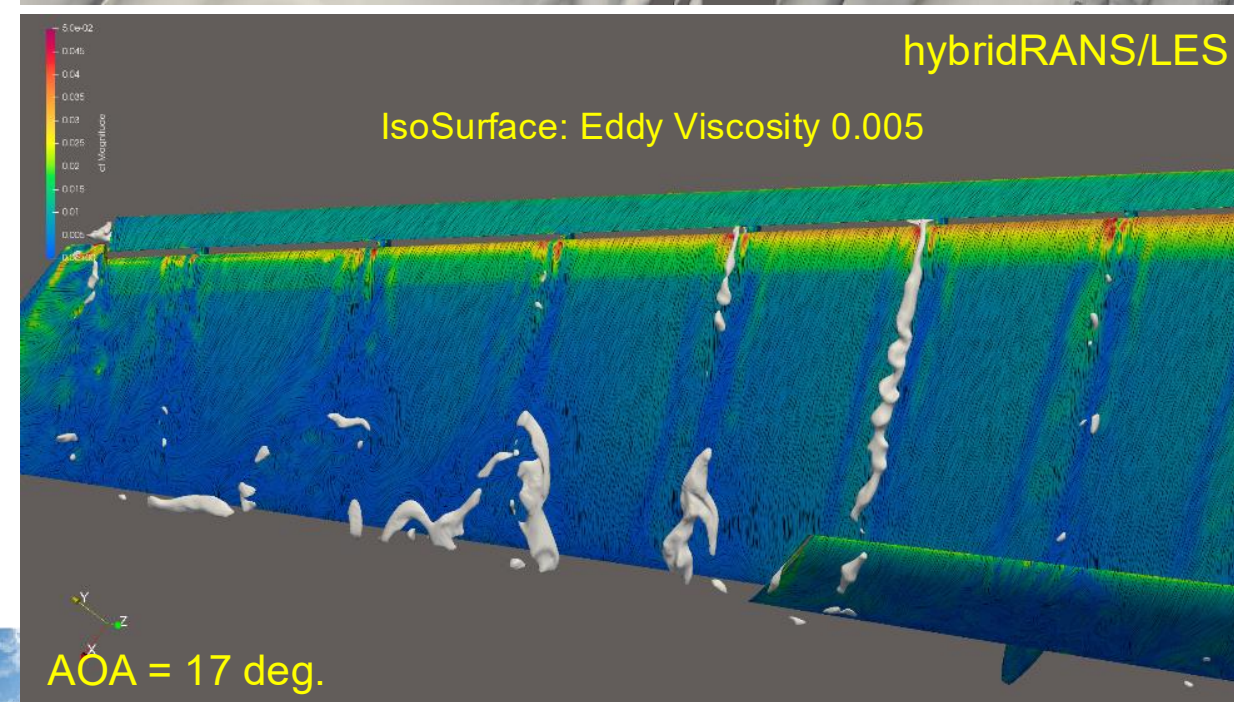
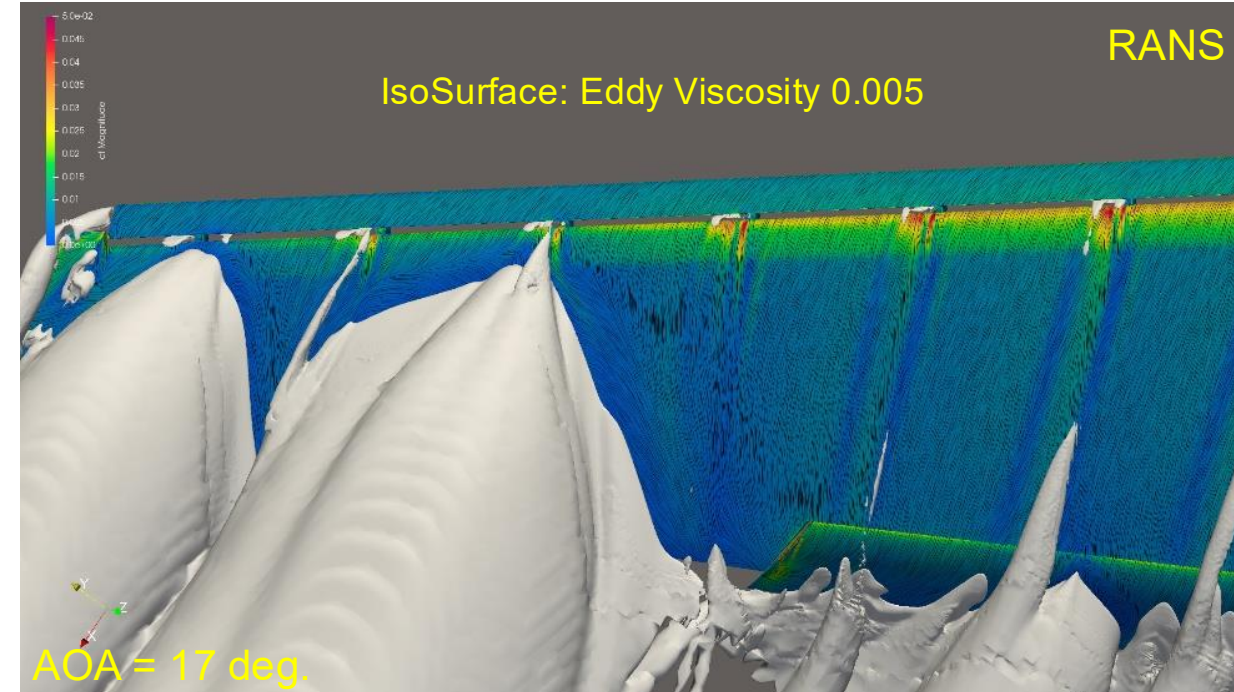
Blockage of the main flow by separated region instead of flow “through”:

- suppresses momentum exchange compared to uRANS & HybridRANS/LES & experiment
- acts like an “obstacle” → additional load on boundary layer upstream of the separation
- drives its upstream growth significantly
- quite fast process in small AOA range compared with uRANS & hybridRANS/LES & experiment



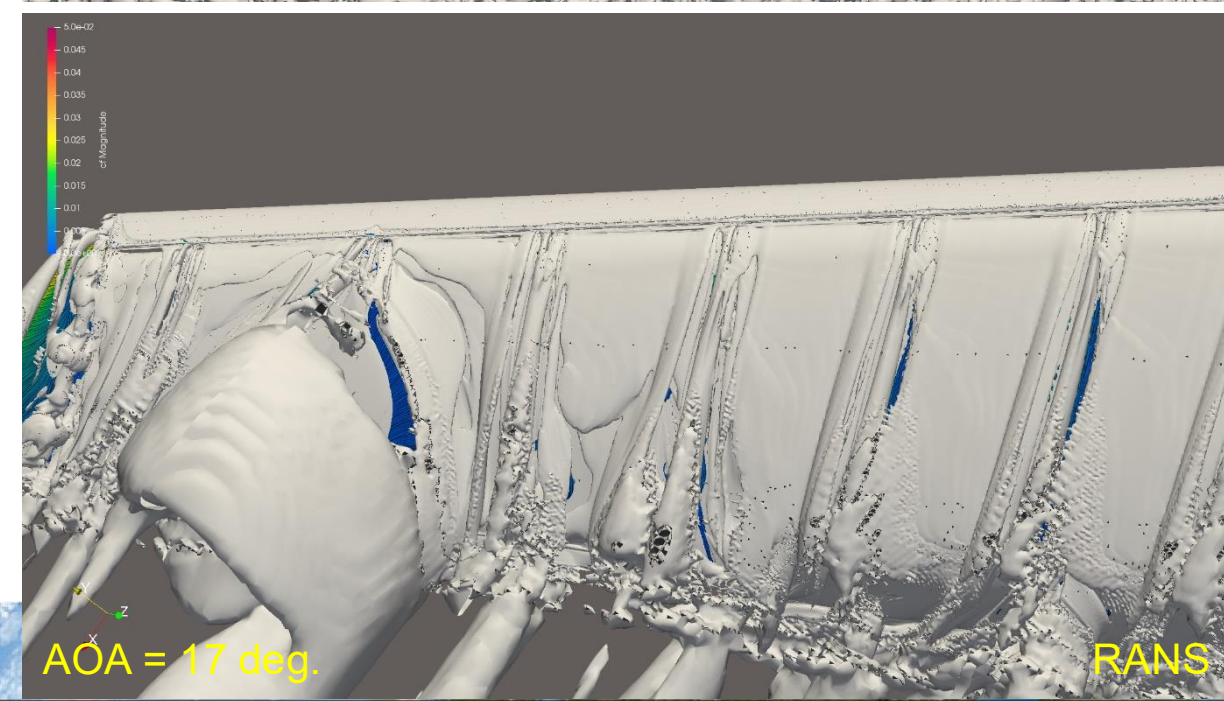
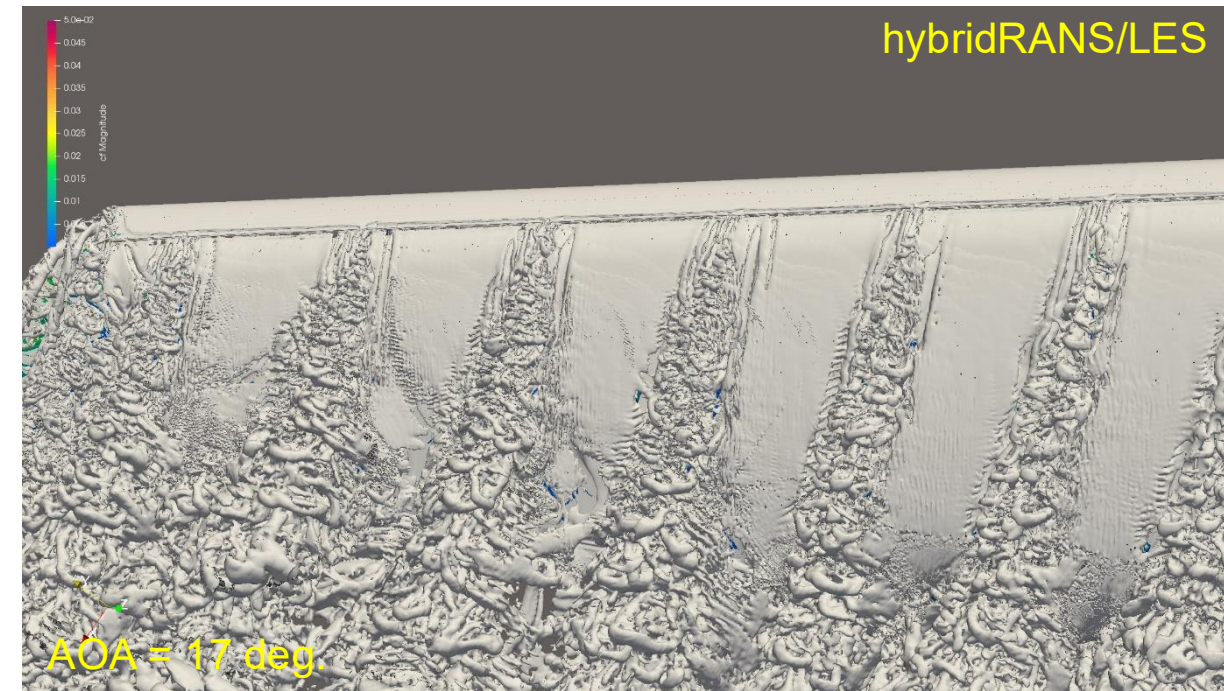
Analysis of RANS / uRANS / hybrid RANS/LES

- RANS: Inside separated region: very high values of eddy viscosity (independent of turbulence model used) → Reason?
- (Expected) input to turbulence model: spatial gradients of velocity (dV/ds)
- Time dependent flow: velocity gradients in space and time (dV/ds & dV/dt)
- uRANS: dV/ds and dV/dt are separated by numerical scheme (dual time step) and time steps are small (global time step): only dV/ds → turbulence model
- RANS & time dependent flow: $dV/ds + dV/dt$ → TU-model:
 - takes the sum of both gradients as “ dV/ds ”
 - additional production of eddy viscosity because of dV/dt
 - damping of time dependent fluctuations
 - balance between dV/dt and eddy viscosity at high levels
 - “converged” solution possible with RANS if balance is stable
- Dependency of dV/dt from time step & CFL: CFL up → $dV/dt \gg dV/ds$
 - different balance point for strong implicit / weak explicit solvers?
 - different “steady” solutions?
 - dV/dt : stability problem for strong flow solvers & high CFL?



Analysis of RANS / uRANS / hybrid RANS/LES

- RANS timestep calculation → local time stepping based on CFL condition:
 - Steady state flow: time depended fluctuations occur only in transient phase of convergence → local time stepping does not change the converged result
 - Time dependent flow: physical dispersion of a time dependency in time & space → “smoothing out” of any time dependent structures:
 - as stronger the type of time integration
 - as higher the CFL number
- CRM-HL “pizza slice” separation due to RANS:
 - Local time stepping → treating local time dependent flow as “steady state flow”
 - uRANS or HybridRANS/LES resolving time dependent flow → physical solution of the problem

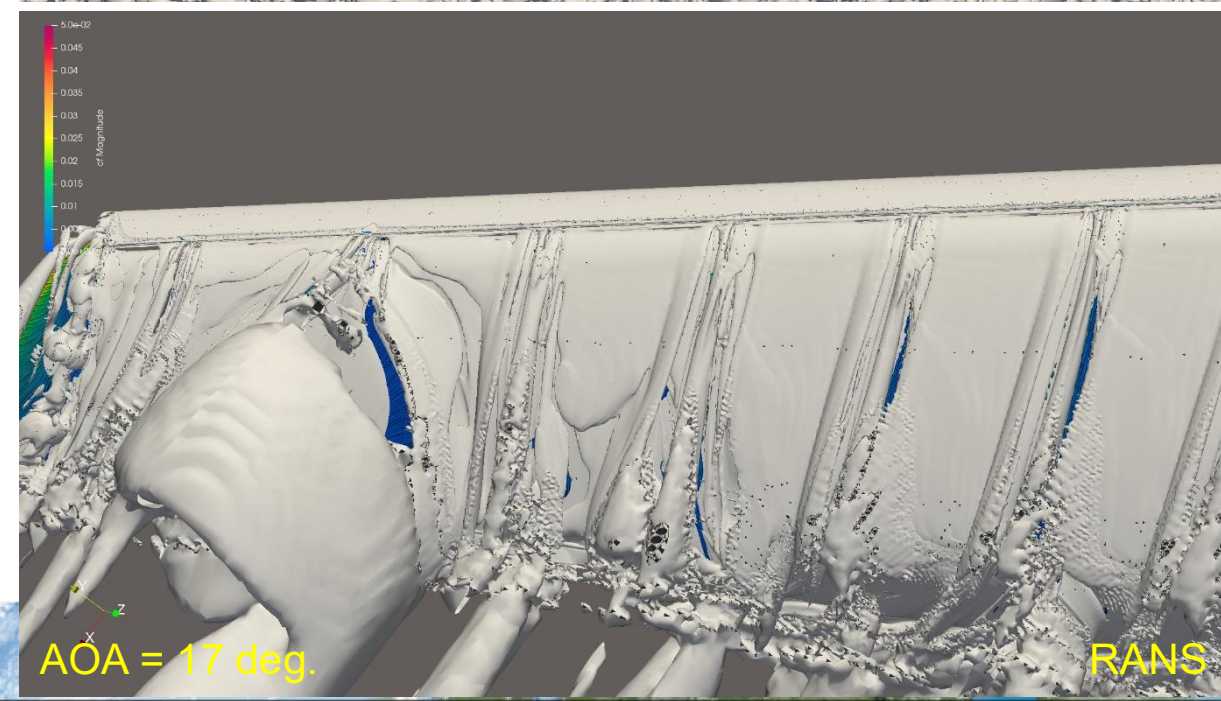
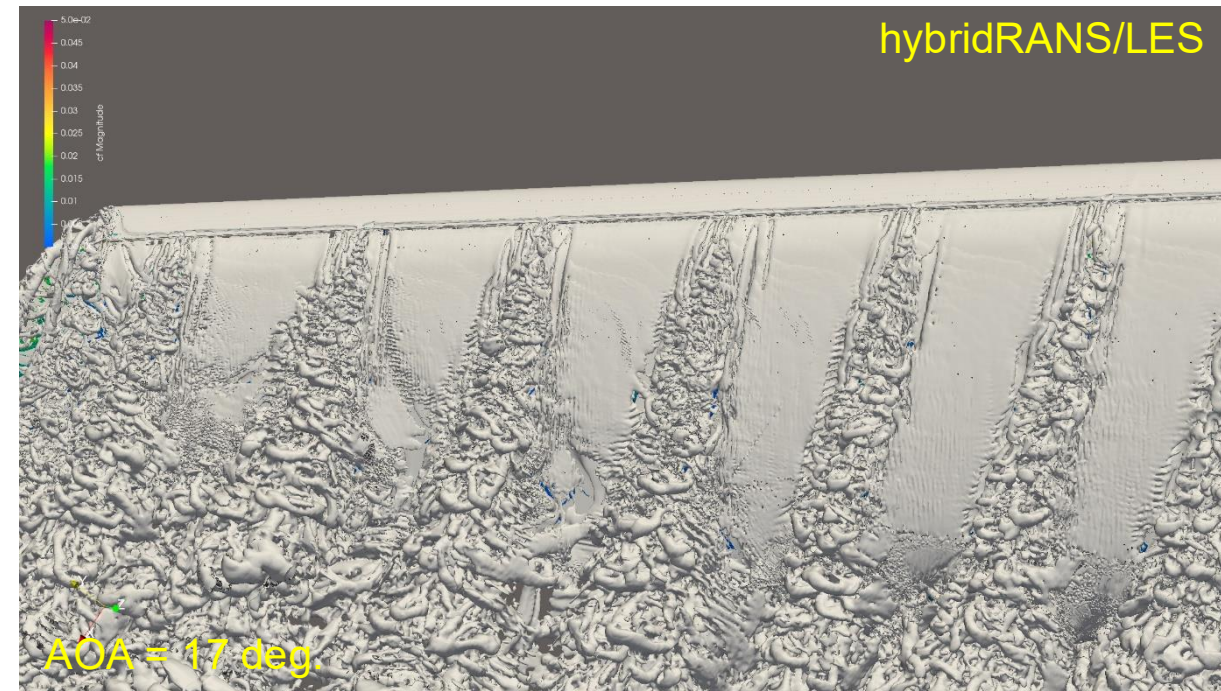


Analysis of RANS / uRANS / hybrid RANS/LES

- Objection: But what if a RANS flow solver is able to converge to machine accuracy including lambda-shaped separations of “pizza slice” type – is this solution “physical”?

A RANS converged solution of an mixed steady/unsteady flow field is an artificially forced steady state interpretation of an overall physically time dependent flow field → result is not physically relevant as soon the time dependent flow influences in any way the global solution of the problem

- This holds true for any steady state RANS solution if local time dependent regions occur and so the findings here are not limited to high lift flows with lambda-shaped separation patterns – they are true for any flow with partial or full time-dependency



Development of xRANS-Method



Development of xRANS-Method

- Main result: all local time dependent flow needs to be treated time dependent → unacceptable costs for typical industrial applications or numerical optimization
- (Try to) reduce costs to level of a steady state RANS: a new method is proposed → extended RANS or xRANS
- Main difference between RANS and uRANS is treatment of the time step for each cell:
 - RANS: local (variable) time steps → just a mathematical tool - physical time is meaningless here
 - uRANS: global (= physical) time steps → time advances uniformly, events happen at their physical rate
- **Basic idea of xRANS: identify domains of time dependent flow & limit time step in this regions to uniform global time step → fully automatic switching from RANS to uRANS in regions of time dependent flow**
- Identification of time dependent regions: using already available variable in RANS solvers → eddy viscosity → high values in time dependent regions due to dV/dt
- Upper time step limit:
 - free parameter of xRANS
 - needs to be chosen depending on numerical and physical characteristics
 - can be possibly “some “order of magnitudes larger than physical time step needed to resolve the time dependent flow – ongoing work ...
 - needs just to be small enough to prevent “RANS behavior” in such regions → possible substantial advantage regarding wall clock time for xRANS compared to uRANS
 - only limitation to an upper limit to prevent numerical instability for small cells in such regions
- xRANS: freely tunned between RANS (no limitation) and uRANS (limitation to physical time step size) → new tool for numerical simulation of mixed steady and time dependent problems



Development of xRANS-Method

- Current implementation of xRANS (in TAU-Code):
 - First loop for all Cells:
 - If time step > xRANS time step limit $\rightarrow$ time step = xRANS time step limit
 - If time step > maximum time step $\rightarrow$ time step = maximum time step ($\sim 10^3$ to 10^4 of xRANS time step limit)
 - Limit the overall time step range – the smaller time steps limit the overall convergence in any case
 - Possibly increase CFL number without instabilities
 - Second loop for all Cells:
 - Time step smoothing by given factor of maximum change between neighboring cells (e.g. 1.2 – 1.4)
 - Smoothed time step distribution seems to stabilize the convergence
- Initial tests: starting from already separated RANS solution $\rightarrow$ switching to xRANS $\rightarrow$ not reliably remove the “pizza sliced” separations
- Requirements for a successful application of xRANS:
 - Start xRANS from a non-separated RANS solution & increase the AOA then by moderate steps – or even better: use xRANS for complete lift curve because of its automatic switching between uRANS & RANS
 - Reliable mark all time dependent flow features $\rightarrow$ eddy viscosity
 - Choose a reasonable time step limit
- Some solutions of the slat bracket problem “just with RANS” to be reviewed based on the ideas of this talk:
 - (excessive) grid adaptation or initially significant refined grids: small cell volume & possibly even constant cell volume $\rightarrow$ time step is limited because of the CFL condition $\rightarrow$ global time step $\rightarrow$ uRANS in these regions
 - time step smoothing (increase convergence stability for complex problems - meaning in many cases – locally time dependent): limits the maximum change of the time step between two neighboring cells to a given factor $\rightarrow$ small factor $\rightarrow$ kind of uRANS behavior

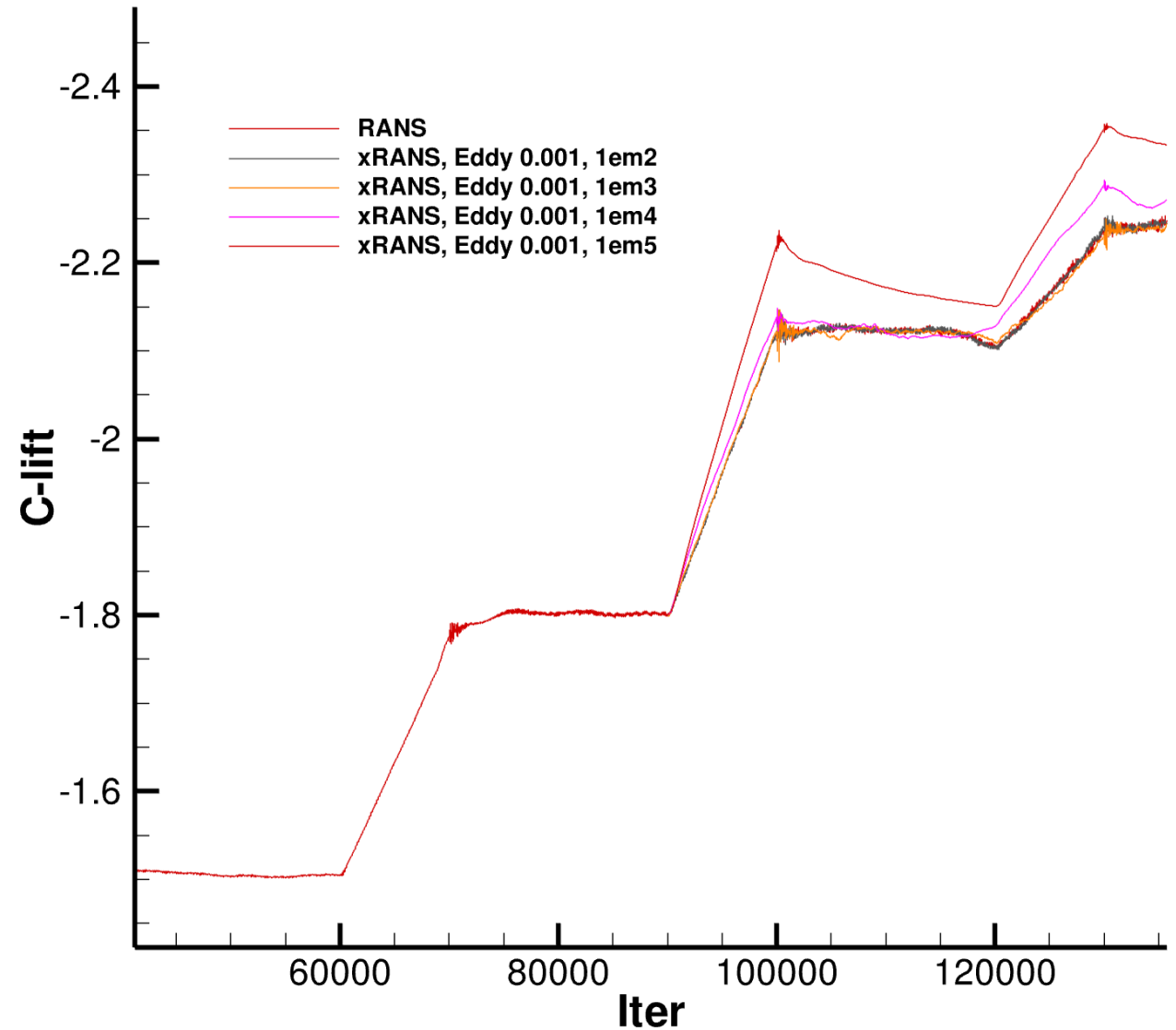


Demonstration of xRANS- Method, Results and Discussion



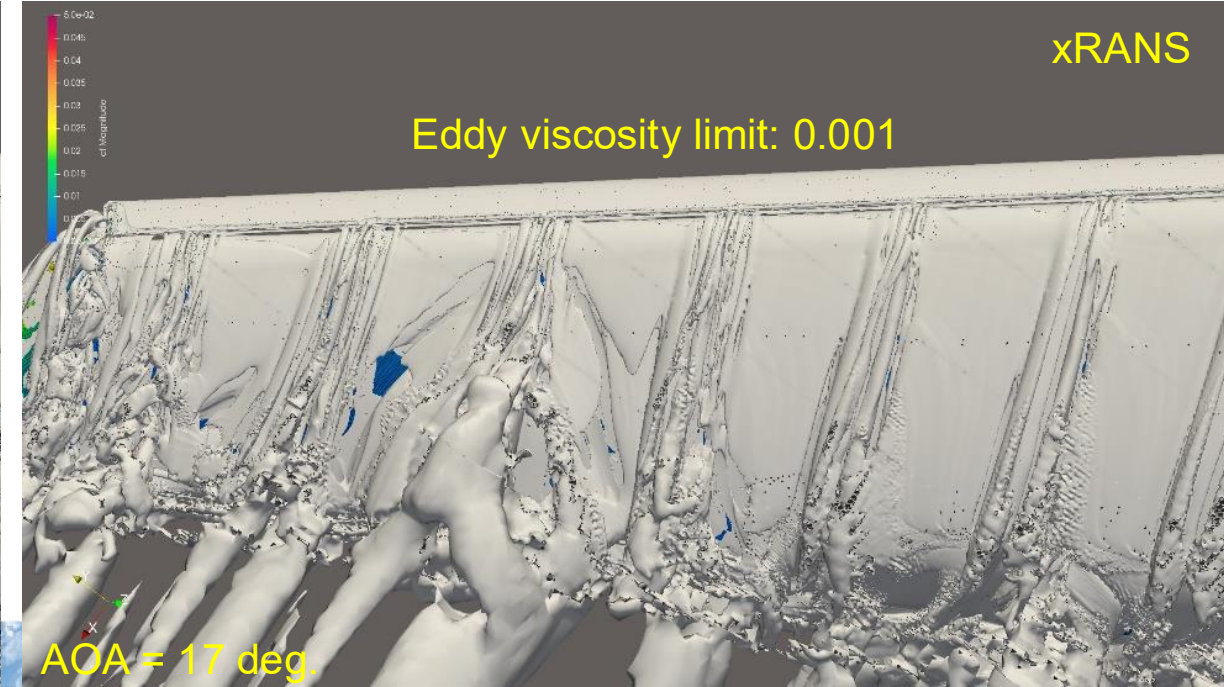
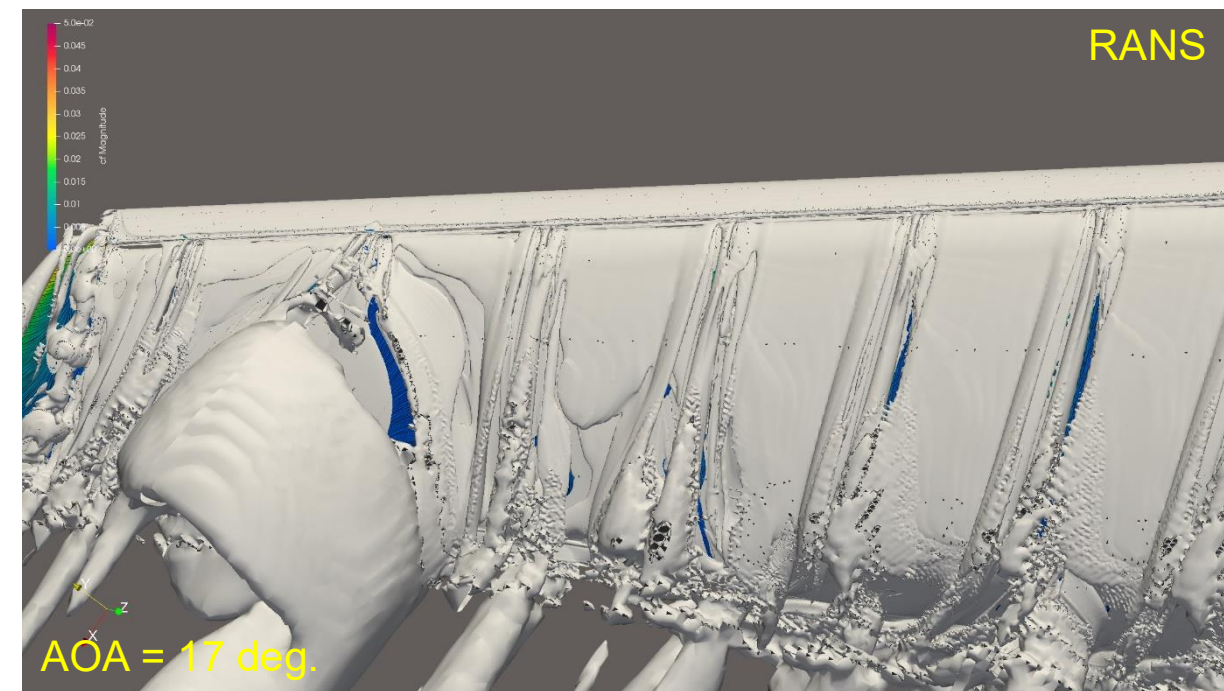
Demonstration of xRANS

- All presented results: Work in progress ...
- Best practice RANS lift curve: Prevent any kind of possible flow hysteresis
 - cold start at AOA 5 degrees
 - pitching up by warm start of the previous AOA
 - steps between simulated AOAs reduced from 3 till finally 0.5 degrees in the range of maximum AOA
 - AOA changes: smoothly variation with small increments ($\sim 1/10000$ degree) by rotating the mesh block with CRM configuration in background mesh (chimera)
 - Same as in experiment
- Time step / Time step limit:
 - uRANS: 0.004226 sec
 - hybrid RANS/LES: 0.000169 sec
 - xRANS: 1.284 - to 0.00128 sec
- Eddy viscosity limit:
 - 0.005 is based on analysis RANS at 17 deg. AOA.
 - Later on: 0.001 – because 0.005 did not mark “all” time dependent areas in full detail



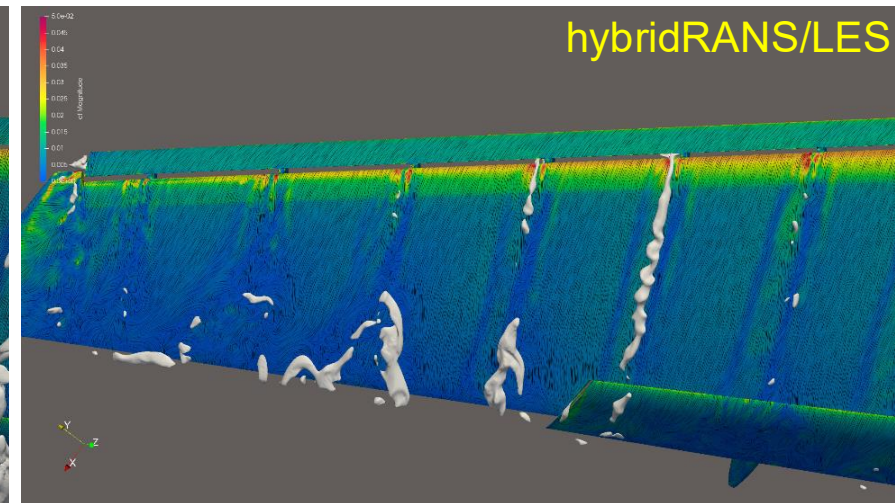
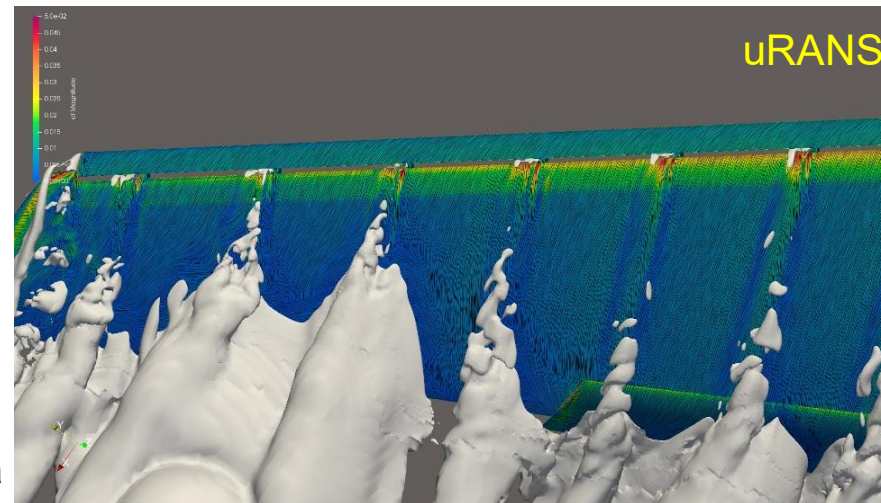
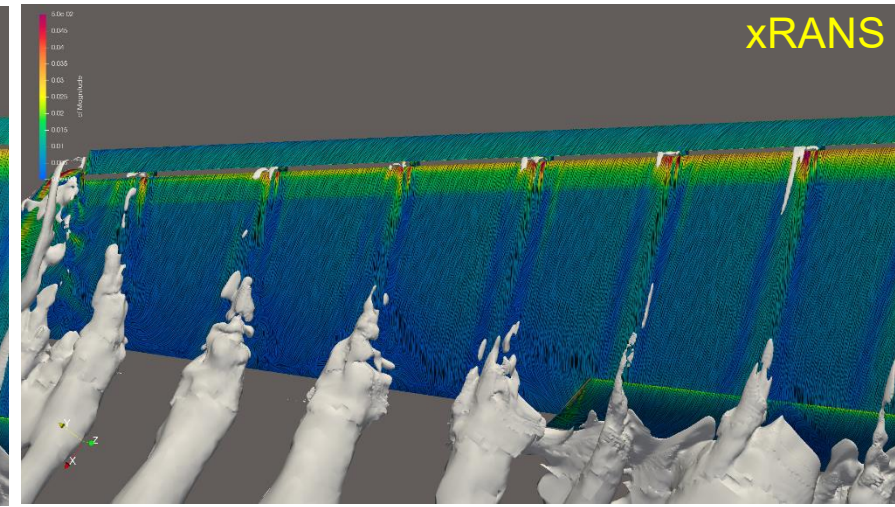
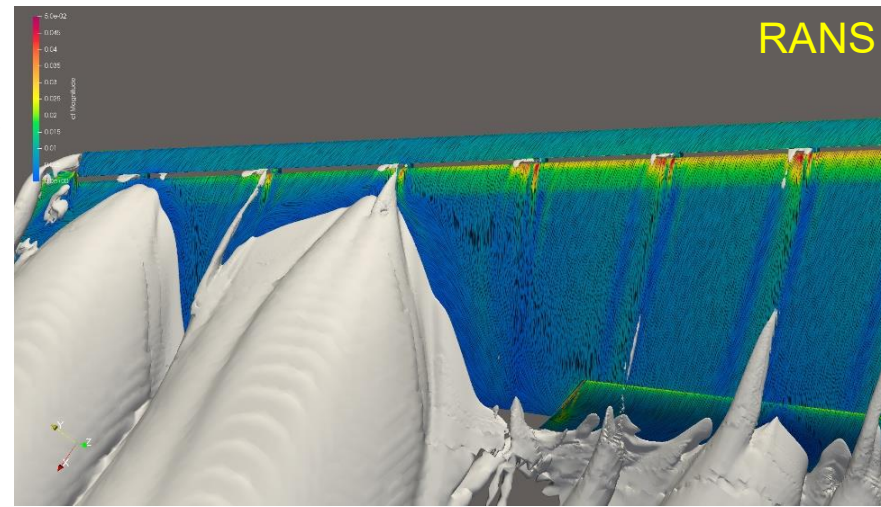
Demonstration of xRANS

- Even with “very big” time step limit of 0.0128 sec compared to uRANS time step $\rightarrow$ xRANS improves RANS results significant $\rightarrow$ clear demonstration of time dependent features are the reason for the “pizza slices”
- Remark: xRANS limited points: 4 – 13% of the grid



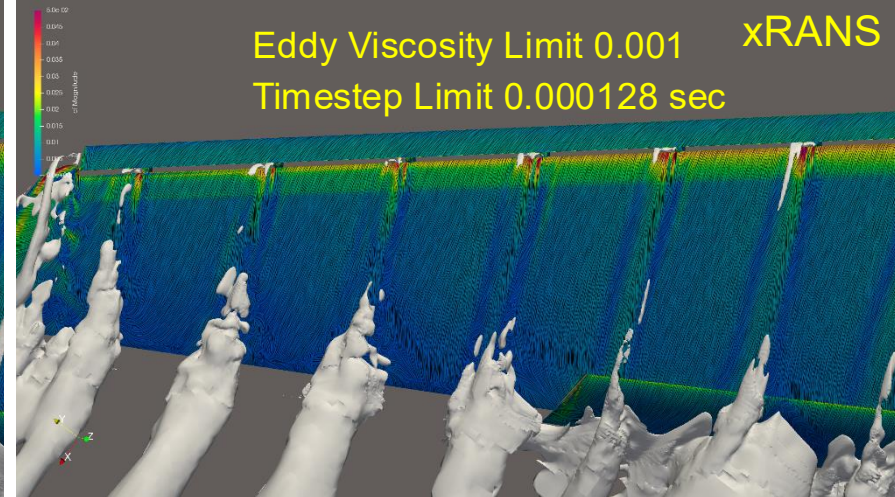
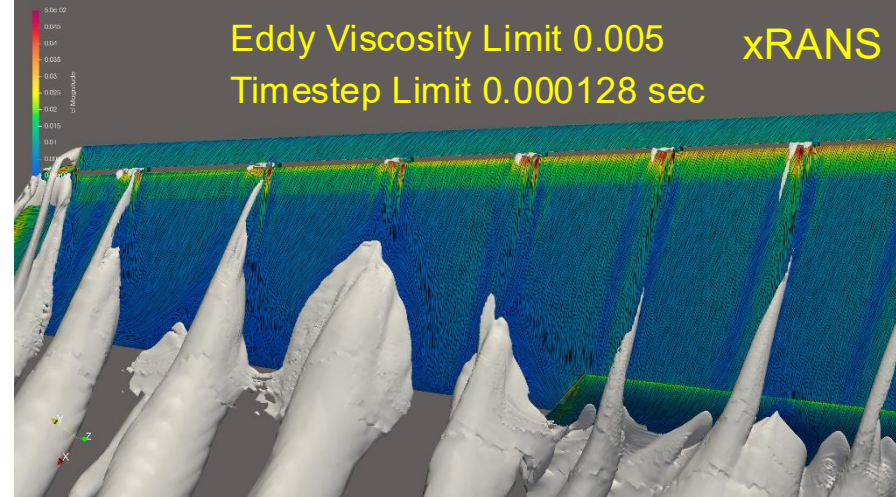
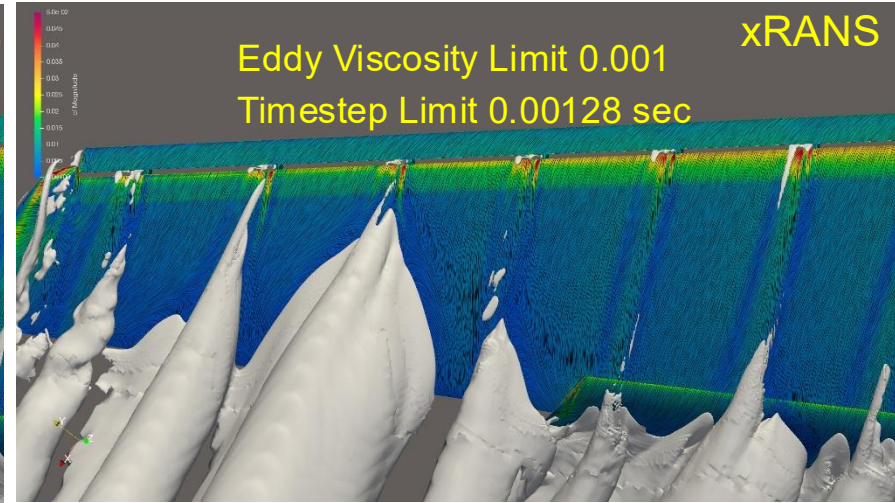
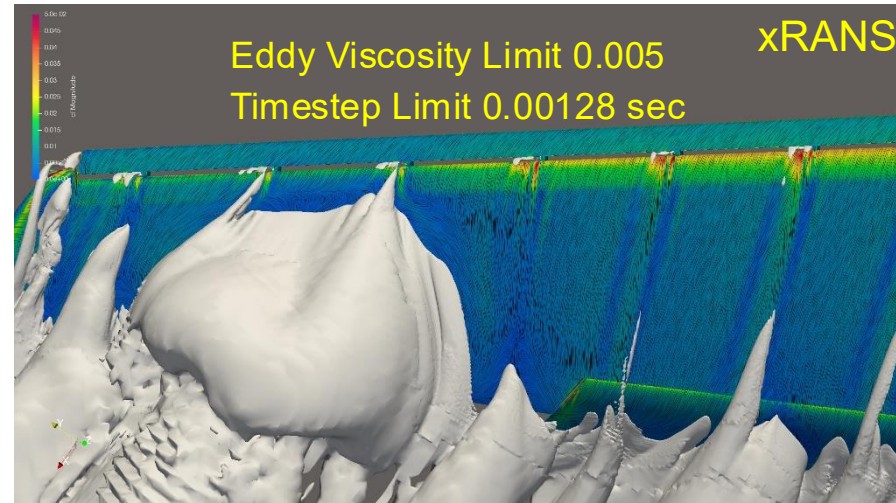
Demonstration of xRANS

- Visualization: isosurface of eddy viscosity = 0.005 → direct effect of xRANS
- Example: AOA 17 deg
- xRANS: time step limit 0.00128 sec, eddy viscosity limit 0.001
- xRANS time step limitation effective to reduce the excessive eddy viscosity production found in RANS simulations down to the same level and distribution as of uRANS → pizza slices “removed”
- eddy viscosity level of an hybridRANS/LES is still lower: – possibly an effect of the LES simulation, but at the same time at a finer grid and a smaller timestep



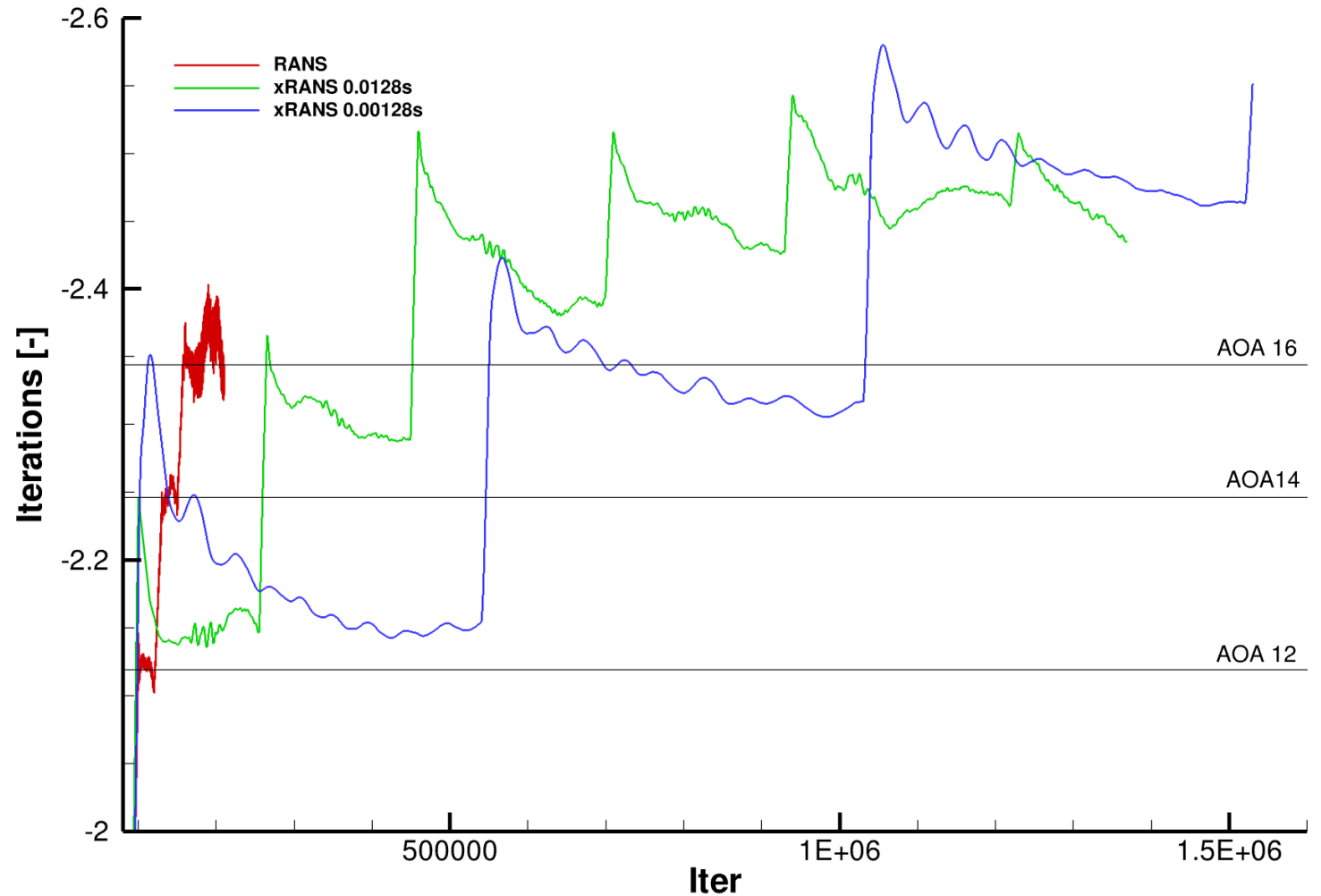
Demonstration of xRANS

- Visualization: isosurface of eddy viscosity = 0.005
- Example: AOA 17 deg
- Variation of time step & eddy viscosity limit
- Stronger effect of time step limit compared to eddy viscosity limit



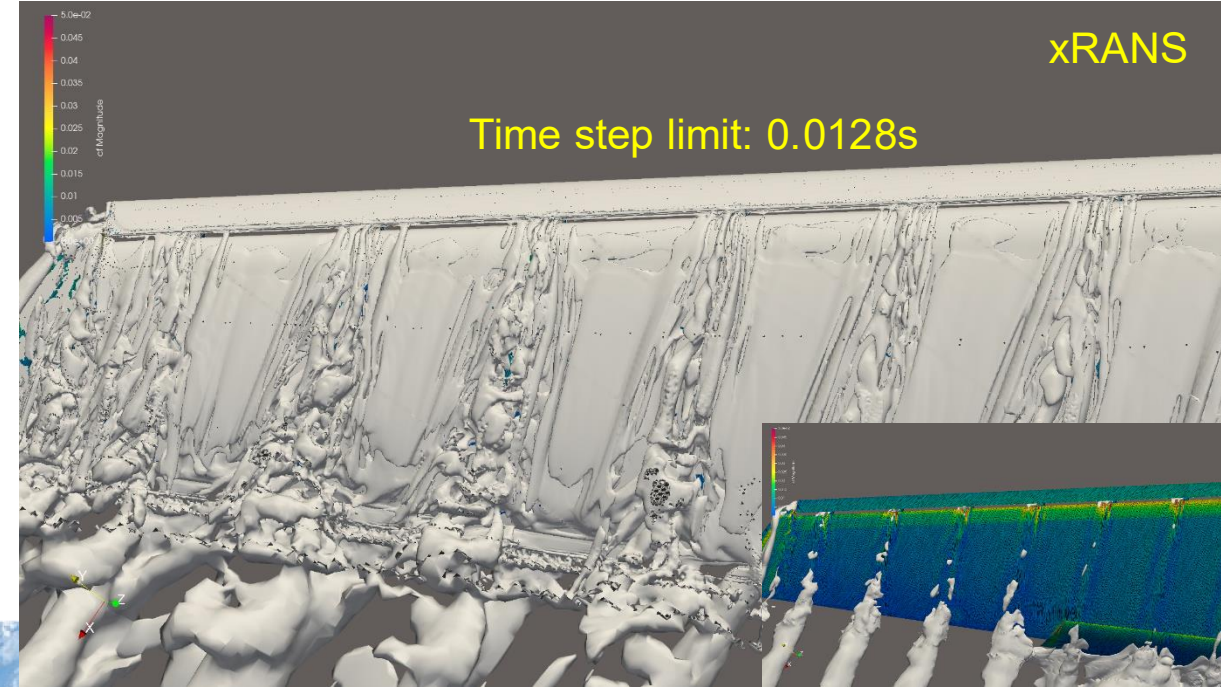
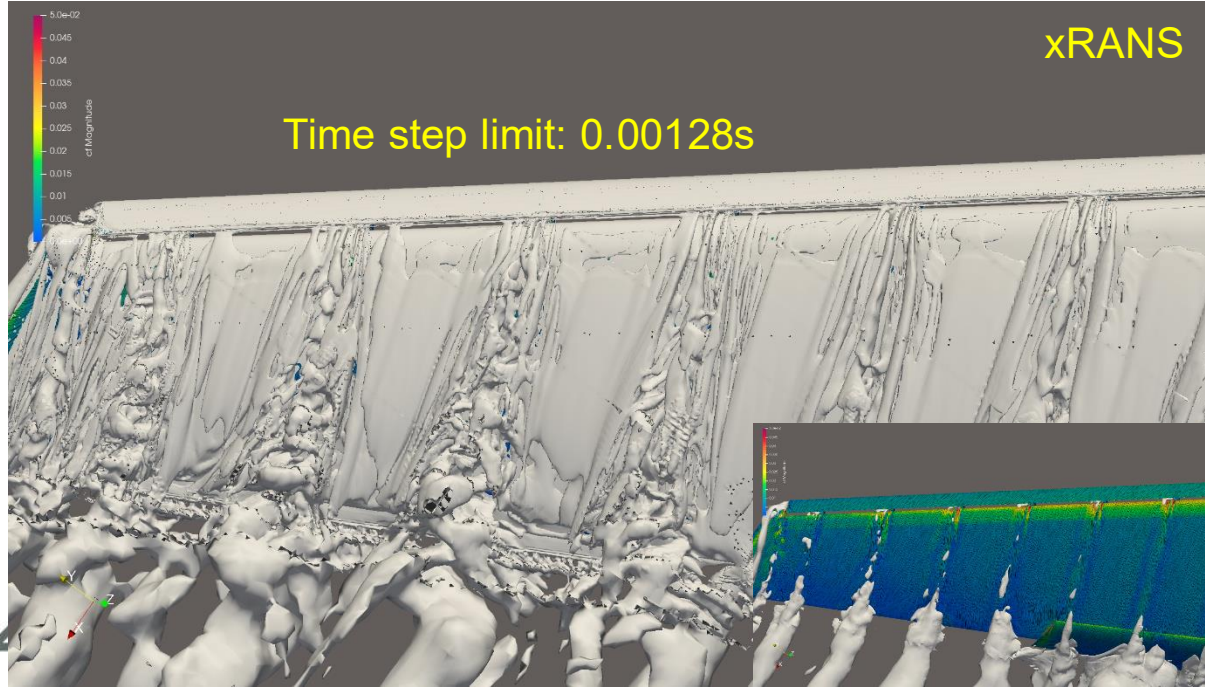
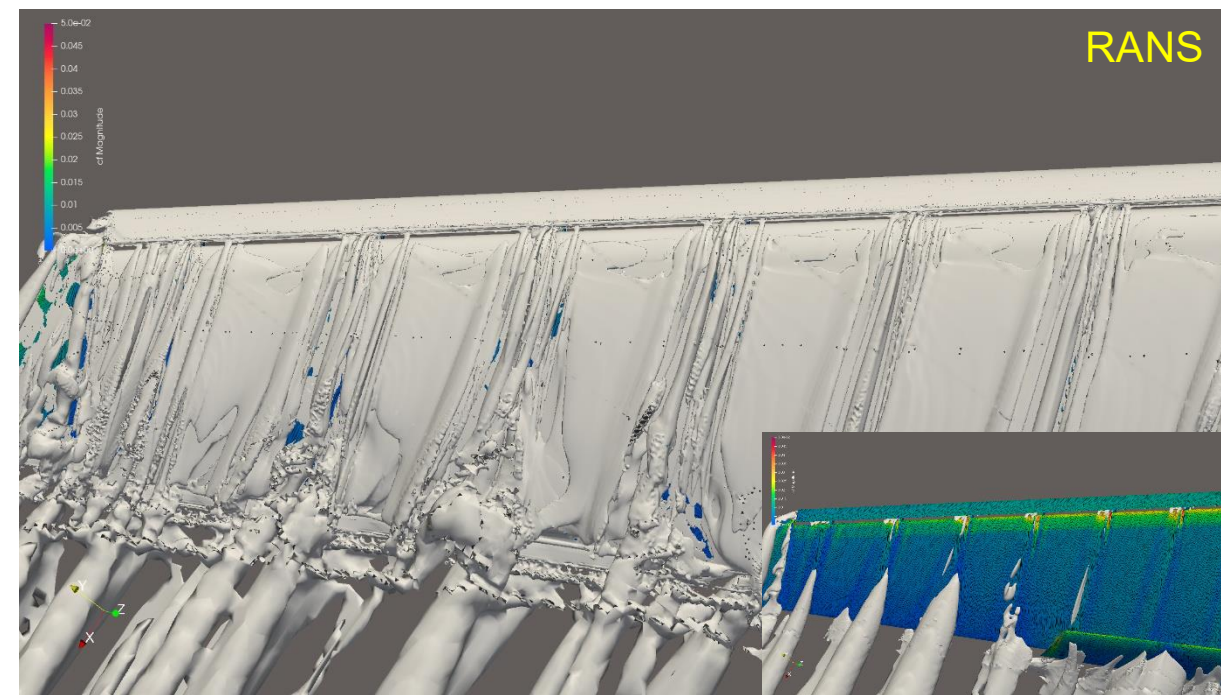
Demonstration of xRANS

- Converged xRANS simulations with current best practice:
 - Eddy viscosity limit 0.001
 - Time step smoothing 1.4
 - Time step limits of 0.0128s and 0.00128s
 - Maximum time step 10k of free stream value
- All polars started from RANS solution at AOA 8 deg.
- Convergence behavior of xRANS turns from “RANS type” to “uRANS type” as smaller the time step limit is → more iterations needed for smaller time step limit
- Overall higher lift level of xRANS results → pizza slices removed – compare next slides



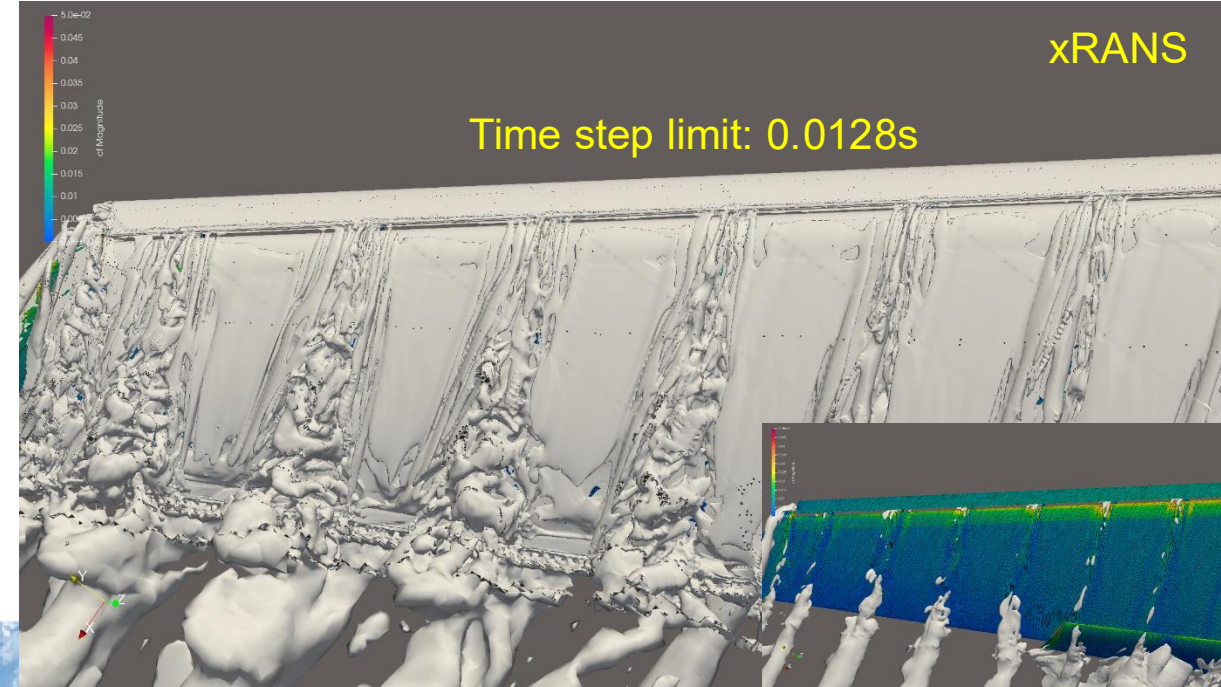
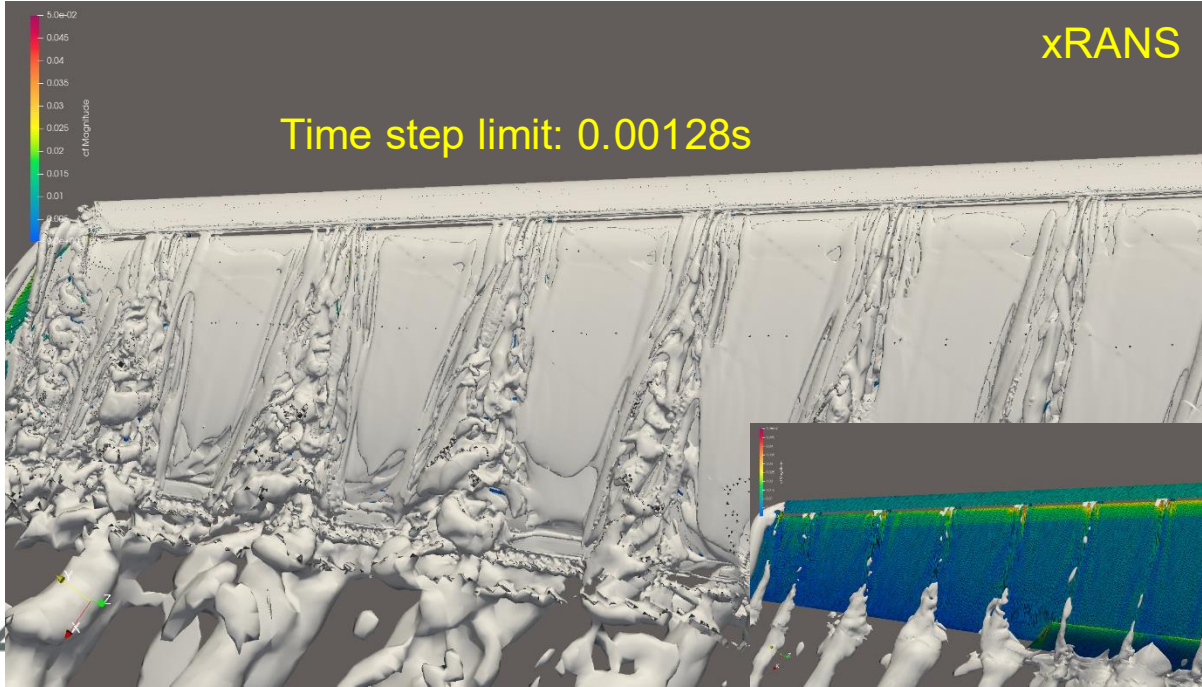
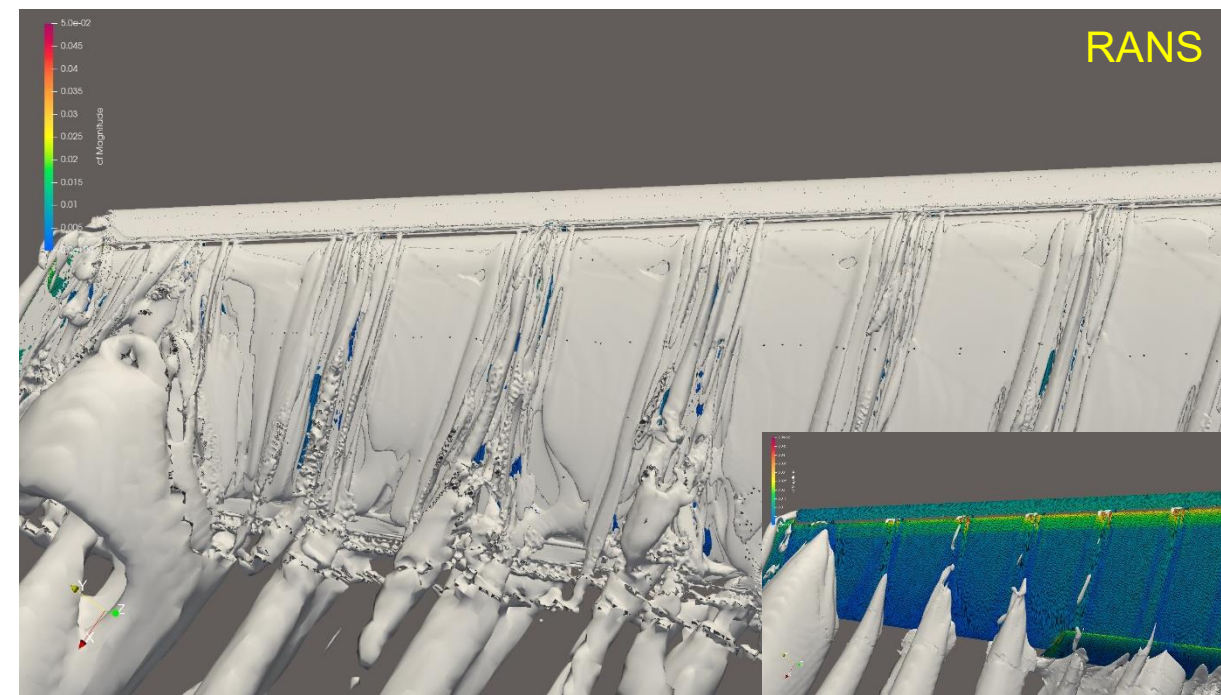
Demonstration of xRANS: AOA 12 deg

- No “pizza slices” at AOA 12
- Much more “vortices” in RANS solution?!



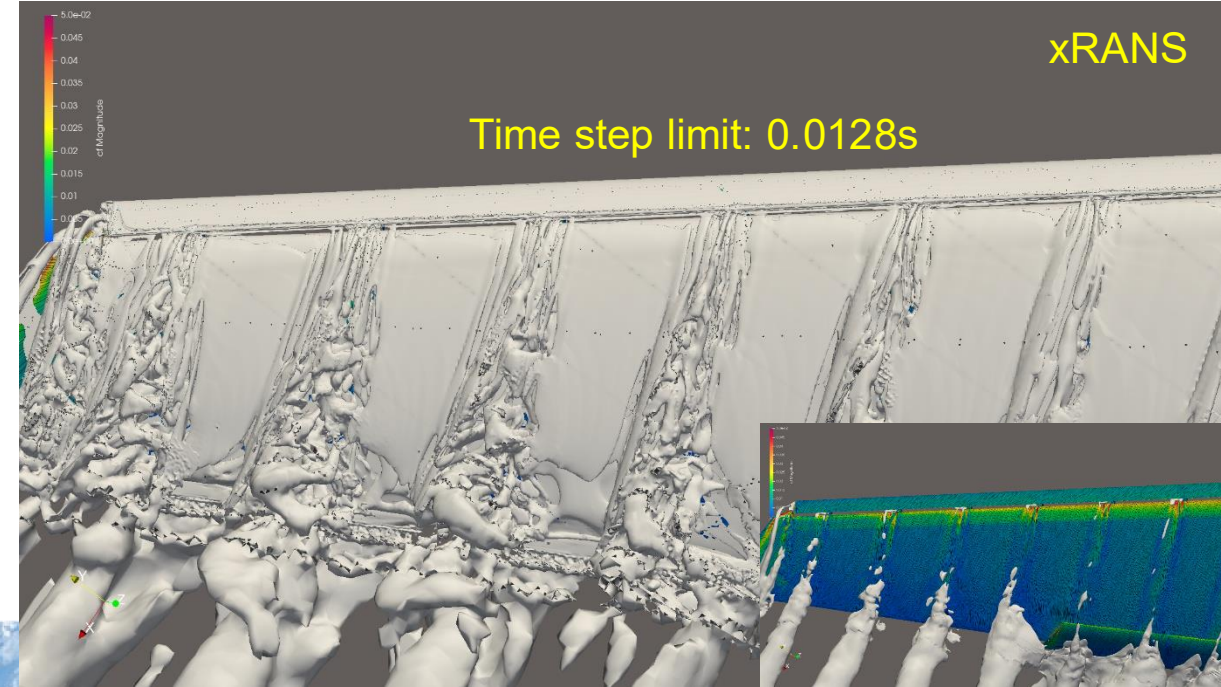
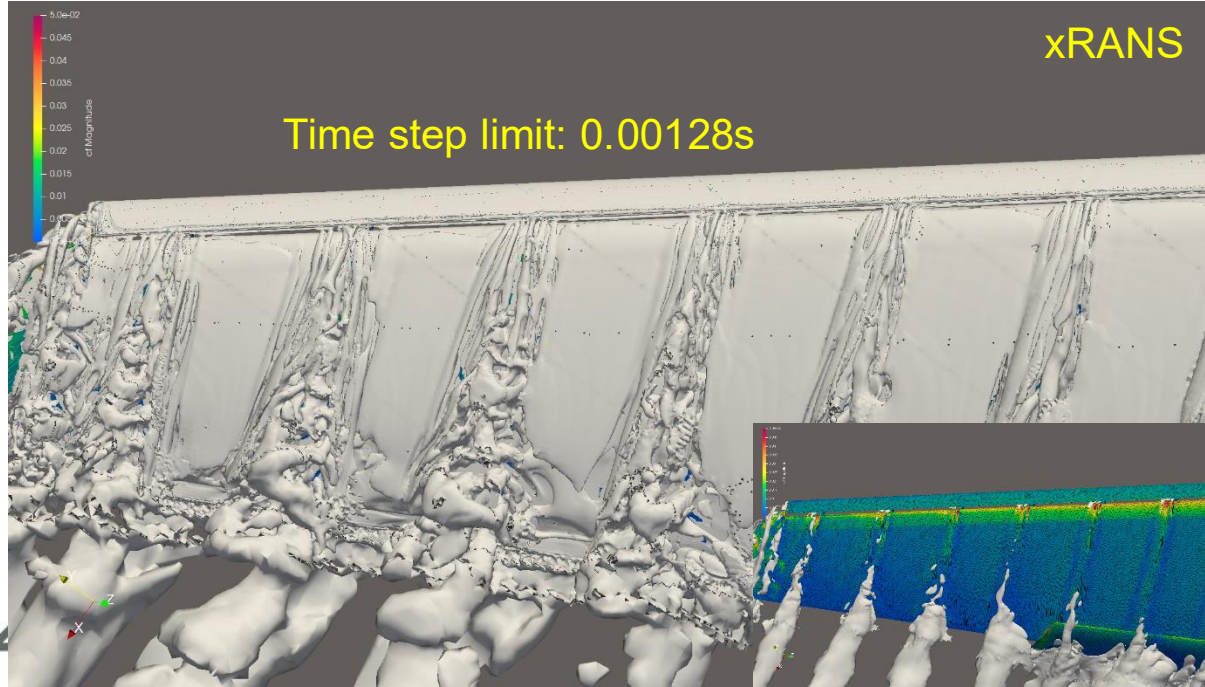
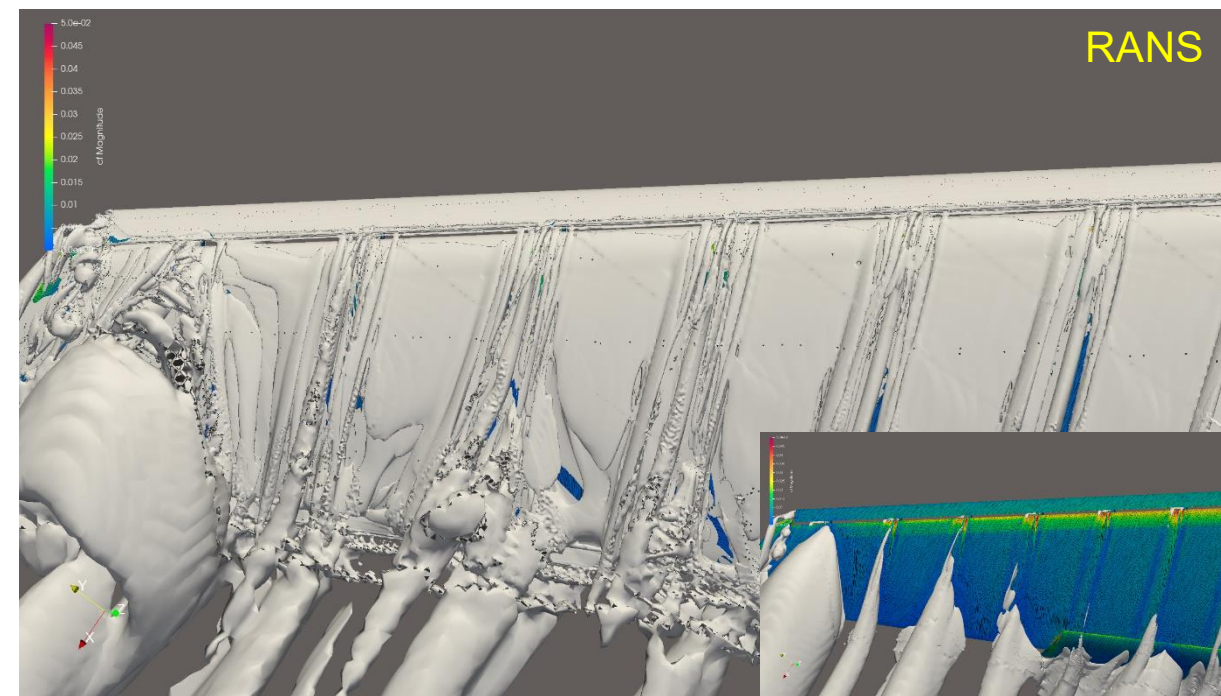
Demonstration of xRANS: AOA 14 deg

- RANS simulation develops pizza slice separation
- xRANS even with time step limit of 0.0128 sec (~ 10x uRANS time step) successfully removes pizza slice separation & flow field has the same features as uRANS



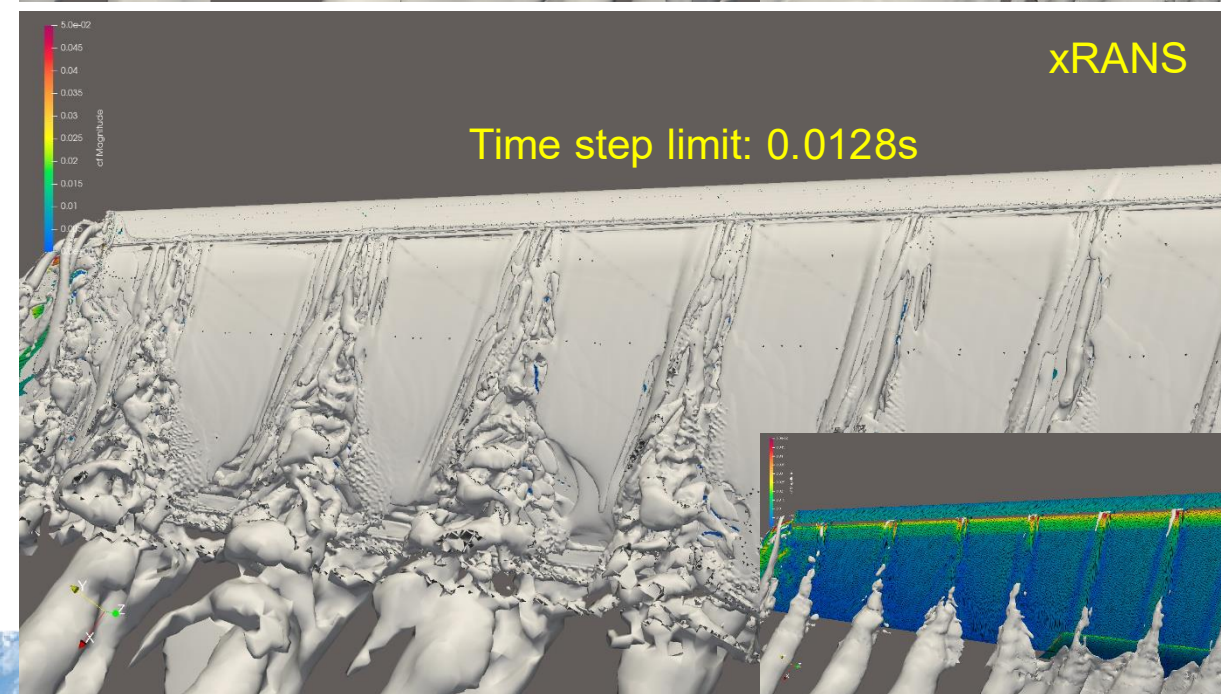
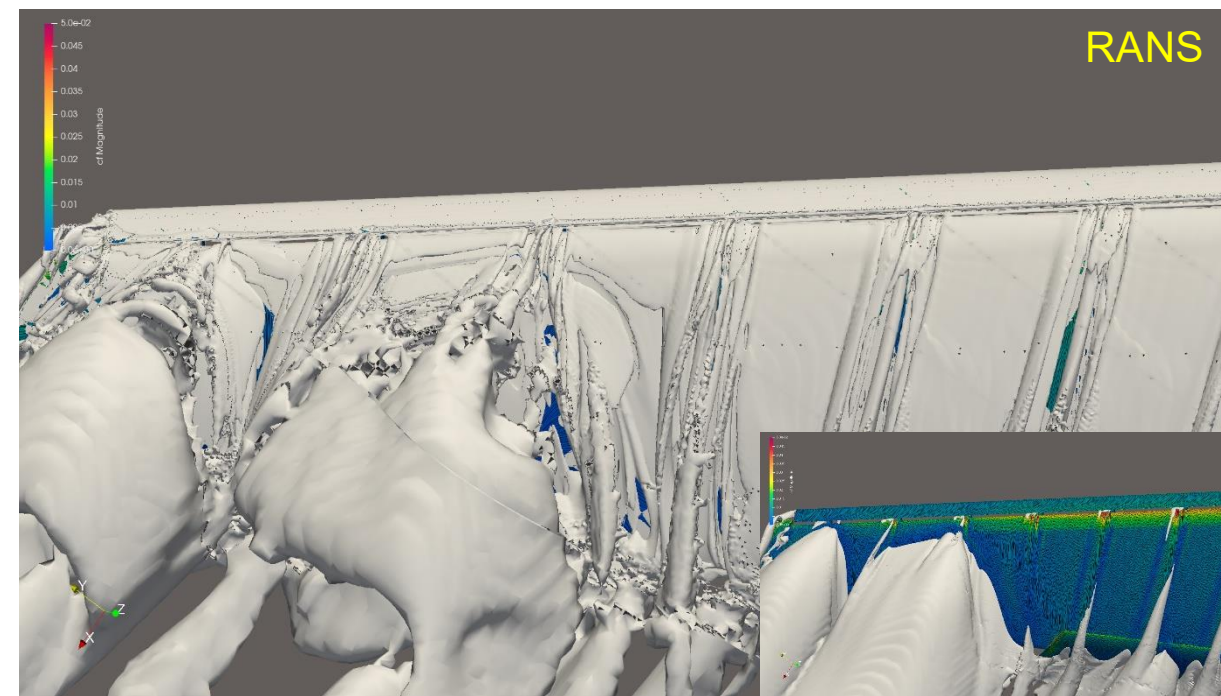
Demonstration of xRANS: AOA 16 deg

- xRANS still without “pizza slice” separation



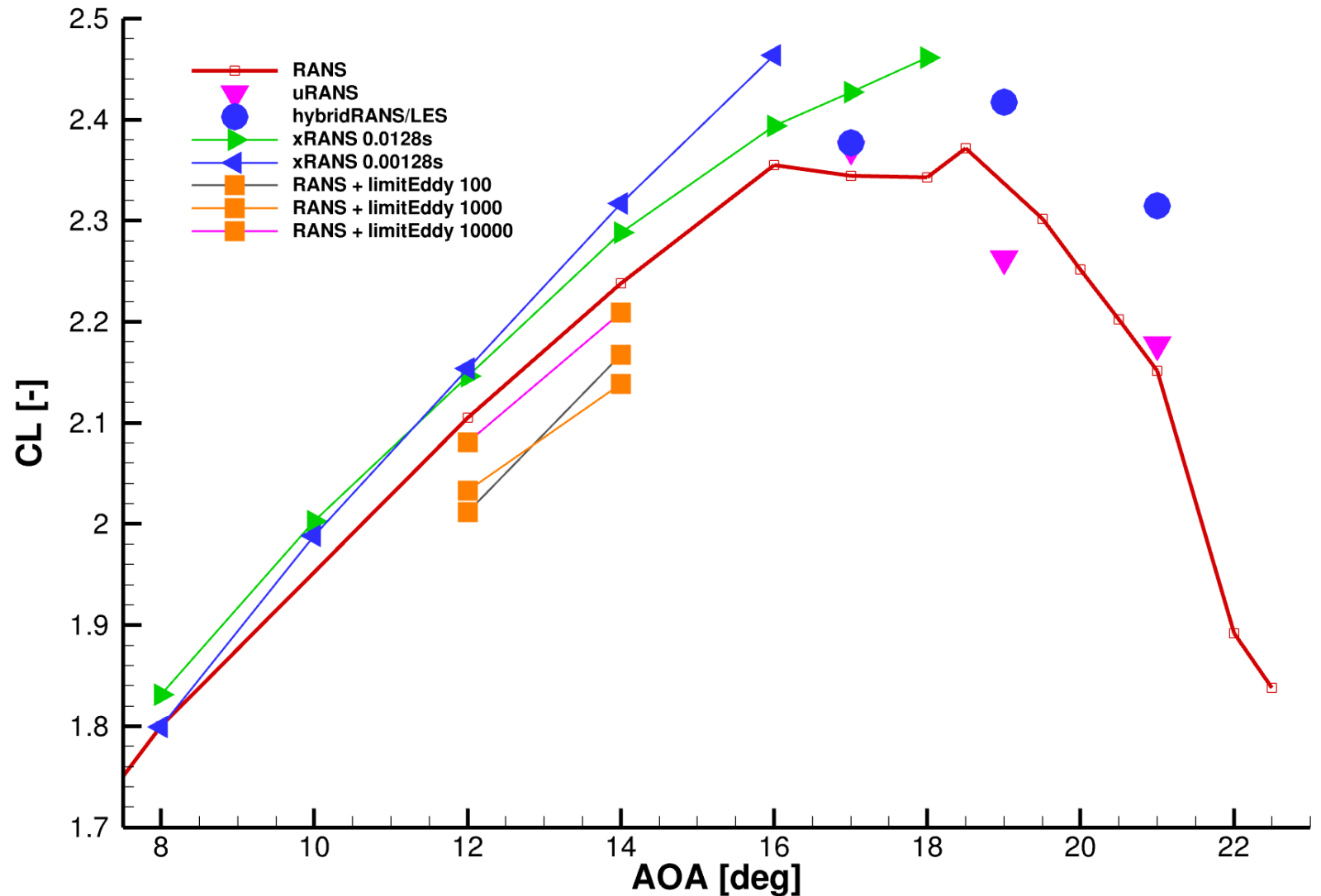
Demonstration of xRANS: AOA 18 deg

- xRANS still without “pizza slice” separation



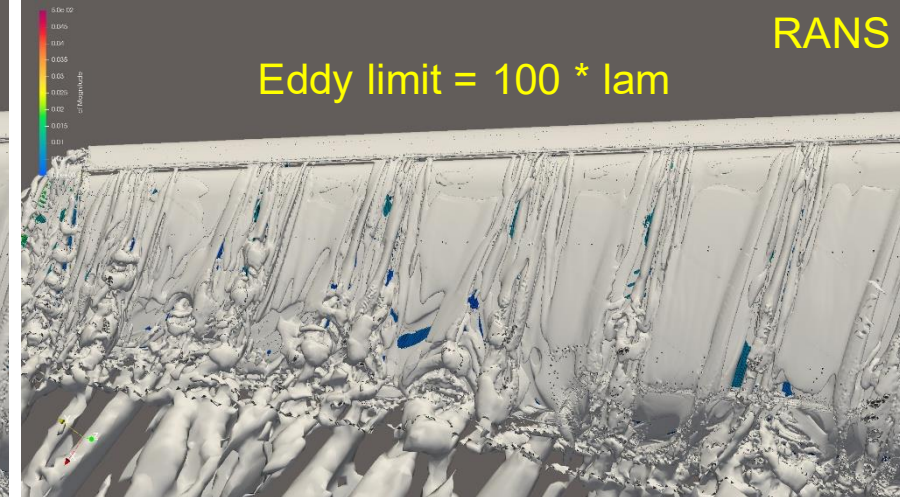
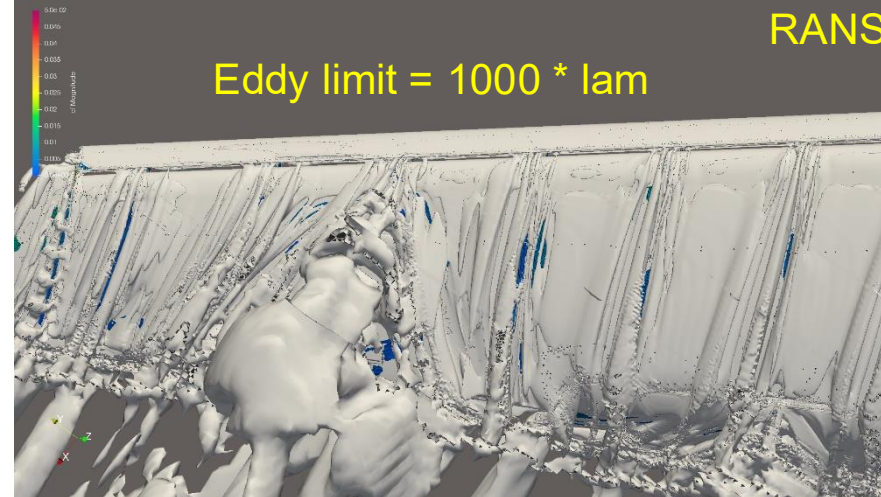
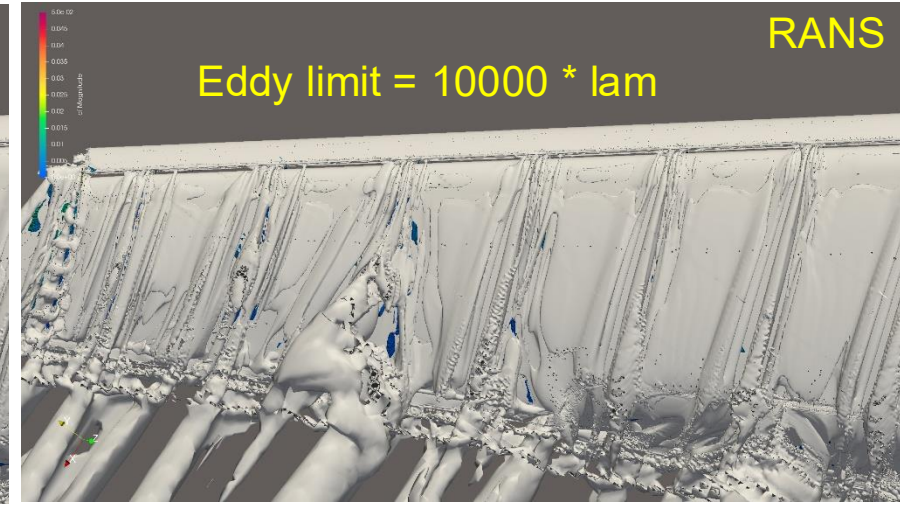
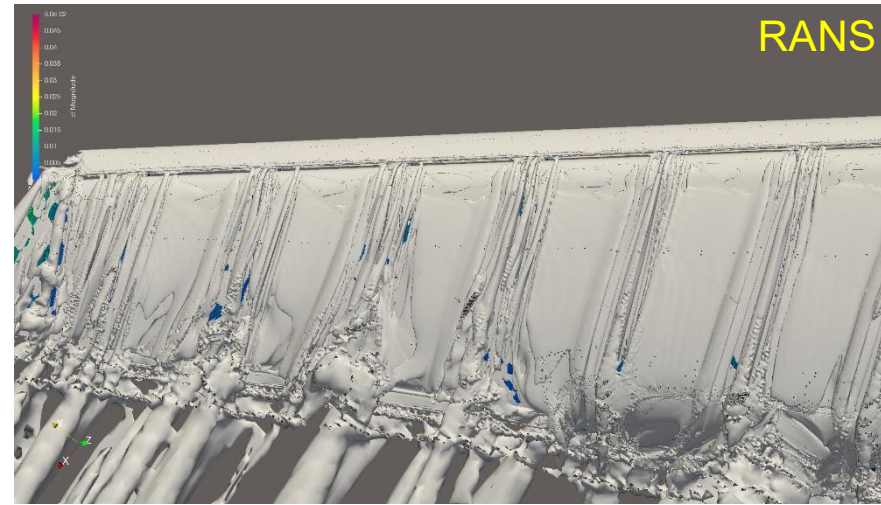
Demonstration of xRANS

- Remark: uRANS at 19 and 21 deg → not recovering from “pizza slices” → new best practice uRANS...
- xRANS: Pizza slice type of separation removed → additional lift
- Additional lift compared to hybridRANS/LES – possible reason:
 - more iterations for xRANS needed because of limited time step → more “CTUs” & mean values
 - Clear effect of time step limit
- xRANS simulations still running ...



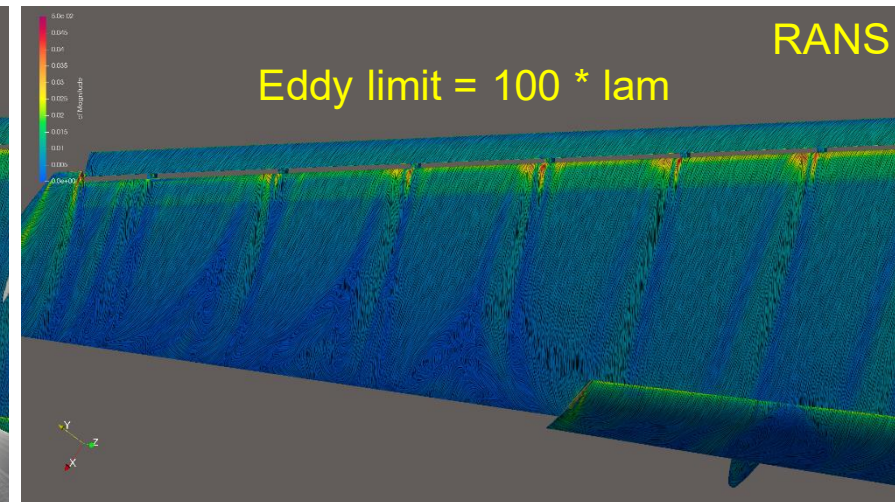
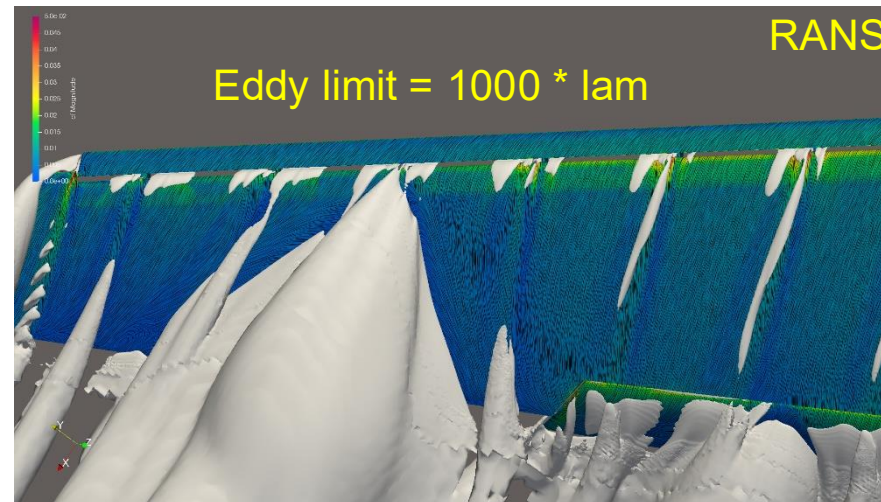
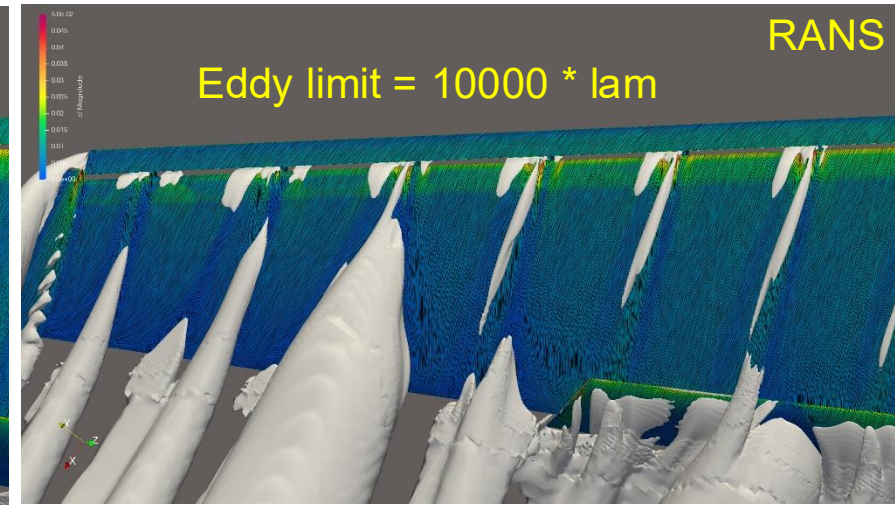
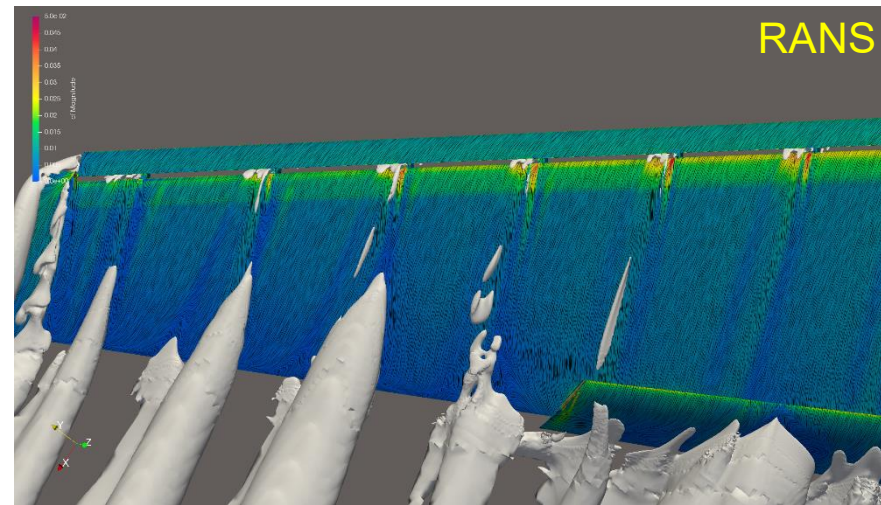
Limit of Eddy viscosity

- Plan B: Instead of using xRANS “just” limit eddy viscosity to a given factor of the reference value in farfield
- AOA: 12 deg
- Pictures: lambda2 criterion
- Limitation to factor of 10k and 1k increases the pizza slice separation – possible reason (much) more iterations in case of eddy limited simulations compared to “pure” RANS → pizza slice creeping up quite slow
- Limitation to factor of 100: seems to create more a kind of LES-result?



Limit of Eddy viscosity

- Comparison of the eddy viscosity in the field: stronger limitation reduces the (not vortex corrected SANeg) results in the vortex cores ...
- Overall:
 - Limitation of eddy viscosity “to small” → pizza slice structures shows up
 - Limitation of eddy viscosity enough to suppress pizza slices → complete flow field is changed - already effect on lift → not useable
 - As more eddy viscosity is limited as more the convergence breaks down → Eddy viscosity limitation creates kind of LES type simulation but without needed small time steps and spatial resolution...



Conclusion



Conclusion

- Comparison of RANS, uRANS & hybridRANS/LES simulation results (HL-CRM): complex flow field with turbulent structures and vortex interaction behind the slat brackets → locally time dependent flow:
 - resolved by uRANS & hybridRANS/LES
 - RANS creates a non-physical solution: Artificial eddy viscosity created by time dependent gradients dV/dt in time dependent regions (independent of the turbulence model)
- Solution of that kind of problem:
 - uRANS or hybridRANS/LES → computational costs
 - xRANS → identify domains of time dependent flow & limit time step in this regions to uniform global time step → automatic switching from RANS to uRANS in regions of time dependent flow
 - Successful demonstration of the xRANS method
- (Current) Open questions:
 - What is the maximum time step limit still getting physical results → walltime speedup compared to uRANS
 - Is there an RANS solution of an unsteady flowfield which is exactly the same as a mean uRANS solution?
 - Fast uRANS-solver – the “better way to go”?
- Paper at SciTech 2027, extended abstract available

